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A SYSTEMS ENGINEERING APPROACH TO ANALYZING WEATHER INPUT SENSITIVITIES OF THE JOINT PRECISION AIR DROP SYSTEM

THESIS

David L. Gemas, Captain, USAF AFIT/GSE/ENY/07J-01

DEPARTMENT OF THE AIR FORCE AIR UNIVERSITY AIR FORCE INSTITUTE OF TECHNOLOGY

Wright-Patterson Air Force Base, Ohio

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AFIT/GSE/ENY/07J-01

A SYSTEMS ENGINEERING APPROACH TO ANALYZING WEATHER INPUT SENSITIVITIES OF THE JOINT PRECISION AIR DROP SYSTEM

THESIS

Presented to the Faculty

Graduate School of Engineering and Management

Air Force Institute of Technology

Air University

Air Education and Training Command

in Partial Fulfillment of the Requirements for the

Degree of Master of Science

David L. Gemas, BA

Captain, USAF

June 2007

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A SYSTEMS ENGINEERING APPROACH TO ANALYZING WEATHER INPUT SENSITIVITIES OF THE JOINT PRECISION AIR DROP SYSTEM

David L. Gemas, BA Captain, USAF

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Abstract

The United States Air Force is partnering with the United States Army as well as allied nations to develop a revolutionary advance in logistical support known as the Joint Precision Air Drop System (JPADS). The focus of this study is to develop a process to quantitatively analyze system sensitivities to various types of weather inputs and the corresponding effect on system accuracy. Weather balloons were used to provide representative "truth" to which forecast weather could be compared. Each data type was fed into the JPADS Mission Planner to produce navigation points which could then be compared statistically. The process was tested on a limited data set to provide a first look at the variables of forecast resolution and "lead-time." Initial results indicate best system accuracy is achieved for lowest forecast resolution (i.e., 45 km vs. 5 km data) and shortest lead-time (i.e., <12 hrs vs. >12 hrs). This result will not only allow for better accuracy of JPADS, but also reduce bandwidth and transmission time necessary to send weather forecast data to the warfighter.

Acknowledgements

Only one name goes on the front of this thesis, but it could not have been written if not for the efforts and inputs of many people. First, I wish to thank the team of Lt Col Steven Fiorino and Mr Ron Lee, my thesis advisor and sponsor respectively. They've been working this project since the beginning and shared an enormous enthusiasm that was truly infectious. I can honestly say that they made this experience fun! I also wish to express my gratitude to Dr David Jacques for bringing me and this thesis into the Systems Engineering program as a package deal, then letting me run with it.

I am also greatly indebted to one of AFIT's previous graduates, Capt Ryan Eggert of the Air Force Research Laboratory's Information Directorate, Advanced Architecture and Integration Branch. Capt Eggert's branch is working on their own piece of the JPADS pie known as WICID. His programming skills were invaluable and provided the means to greatly increase both the speed and accuracy of data preparation for this study. There was also Mr Bob Holt at Planning System Inc (PSI) and Mr Thomas Fill at Draper Labs. Both of whom fielded a whole host of questions about the inner working of the Mission Planner. Without the efforts of Ms Mary Bedrick, AFWA Det 3, there would have been nothing to study. She patiently answered (and sometimes re-answered) my weather questions. Finally, my many thanks to my AFIT instructors, like Maj Sam Wright and Lt Col Mark Abramson who helped push back the darkness with the light of understanding.

Thank God, it is done!

D. Gemas

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A SYSTEMS ENGINEERING APPROACH TO ANALYZING WEATHER INPUT SENSITIVITIES OF THE JOINT PRECISION AIR DROP SYSTEM

I. Introduction

Background

In the modern world of precision engagement with weapons such as the Joint Direct Attack Munition (JDAM) allowing for accuracy measured in feet, the airdrop community has had to soldier on with low precision tactics and techniques that would be recognizable to their Vietnam era counterparts. The Joint Precision Air Drop System (JPADS) is a family-of-systems developed to bring the sort of precision capability found in Global Positioning System (GPS) guided munitions to the airlift community. As such, JPADS is often touted as "the JDAM for logistics."

The purpose of this thesis is to investigate the impact of weather data inputs on the accuracy of JPADS, specifically the JPADS Mission Planner (JPADS-MP) and the navigation outputs it creates from this weather data. Despite the conceptual similarities between JDAM and JPADS, the airdrop mission poses problems for accuracy not faced in the precision guided munition mission. A JDAM class munition falls at high velocity through the atmosphere and, while guided, still follows a relatively ballistic trajectory. As such, the precision of JDAM type munitions is not greatly affected by the weather they pass through between the launching aircraft and their target. The JPADS chute loads are quite different as they truly do fly as a paraglider. Since they are unpowered, proper energy management during their decent is critical in hitting their designated Point-

of-Impact (PI). In order to achieve the desired level of precision, a Guided Parachute requires knowledge of the state of the atmosphere in which it will fly. This thesis examines the current methods used for weather data ingestion by JPADS and determines best practices for the *as-is* system. It will conclude with recommendations for further research to develop an improved *to-be* system.

Scope

Even as this thesis is being written, JPADS is evolving. At present, it is undergoing the Joint Military User Assessment stage of its Advanced Concept Technology Demonstration (ACTD) testing program at the US Army's Yuma Proving Ground (YPG). Despite the continued evolution, JPADS is already operating in the combat theatre. The fact that JPADS is already in use serves to focus the domain of this thesis. This research is the product of techniques from weather forecasting, operations research, and systems engineering. Any one of these fields can find a rich source of problems for study in the JPADS program. However, there is a pressing, operational question at hand. As the system stands today, in the field, how do we make it as accurate and precise as possible? Other organizations are already pursuing studies of the Guidance, Navigation, and Control algorithms for JPADS. That leaves the question of weather impacts.

Problem Statement

Airdrop operators require an evaluation of the sensitivity of the JPADS-MP to weather inputs. To quantify this sensitivity, it is necessary to first identify what weather

products are used by JPADS-MP and what they are used for. To achieve a manageable scope, this study will focus on two weather products usable by the JPADS-MP: Air Force Weather Agency (AFWA) Forecasts and Weather Balloons. The immediate goal is to statistically compare the various types of forecasts generated by AFWA to actual weather sampled by weather balloons and thus determine the best operational practice. Of even greater value, though, is the process that will be developed to achieve this goal, as it will continue to be useful to the program as an analytical process beyond this initial research.

Research Objectives, Questions, and Hypotheses

Research Objective.

The objective of this study is to analyze the weather sensitivities of the Joint Precision Air Drop System. To do this requires the development of a standardized, statistically sound, method of comparing weather inputs to the JPADS-MP. This research will then use this process to perform an initial analysis to answer the Research Question.

Research Question.

This research will perform an initial analysis of AFWA weather forecasts to determine which, if any, provide better planning accuracy for the airdrop mission when used as input to the JPADS-MP.

Investigative Questions.

- 1. How does weather affect JPADS? (i.e., how does the JPADS-MP ingest and use weather data? What are the outputs?)
- 2. How does the JPADS-MP use weather data to generate navigation outputs?

- 3. How can the navigation outputs from the JPADS-MP be converted to a statistically comparable format?
- 4. What different types of AFWA forecasts are available?
- 5. How are weather balloons made available to the JPADS-MP?
- 6. What statistical tests and tools can be used to analyze the weather sensitivities of JPADS?

Hypotheses.

A key objective of this study is to apply statistics in order to get a quantitative understanding of how JPADS is sensitive to weather inputs. As previously mentioned, JPADS is already operating, both in test and in the field, and is doing so with certain qualitative assumptions about the best practices to use regarding weather inputs. JPADS has already made great strides in accuracy, but more is desired. To get there, qualitative assumptions must give way to quantitative results.

The statistical tests used in this study are relatively simple. In general, the Null Hypothesis (H₀) will be that the given distribution cannot be rejected; while the Alternate Hypothesis (H_a) will be that the given distribution is rejected. All statistical tests in this study will be performed at $\alpha = 0.05$.

Methodology in Brief

A standardized mission will be used in the JPADS-MP, with the different weather products being used to generate navigation outputs. These outputs will then be converted into a common Northing vs. Easting error format which is functionally like the Miss Distance charts used by the JPADS Guided Parachute systems. The data will then be

subjected to statistical examination to determine Goodness-of-Fit for Bivariate Normality (a typical distribution for this type of data). Finally, the means and variances of the different data groups were compared to identify the best weather forecast type for use.

Document Structure

Chapter 2 is a literature review which will provide a more in depth discussion of the topics introduced in Chapter 1. The chapter will begin with an abbreviated history of airdrop then progress to a brief description of the JPADS system, as well as provide some insight into weather forecasting and weather products used by JPADS-MP. Chapter 3 details the means by which the research was accomplished. It includes further information on how the weather data was converted into a useful format for statistical analysis as well as the details of that analysis. Chapter 4 reports the results of the analysis and Chapter 5 provides a summary of the overall research effort as well as presents avenues for further research.

Limitations

The research documented in this thesis is limited to the analysis of historical data rather than a fully designed original experiment. As such, the analysis in this research must use data which was intended for other purposes. In this case, not all potentially available forecasts were recorded. While there are sufficient data points to extract statistical significance, care must be taken in interpreting the results so as not to over generalize beyond what the data supports.

The weather balloons that were used as a basis of comparison for the forecast data were, of course, actually launched to support the aircraft operations of the JPADS ACTD. This makes it necessary to check the data for unanticipated correlations. Additionally, it is worth noting that the primary research question, i.e. the accuracy of various forecast products, was originally raised by AFWA Det 3. It was decided to use the JPADS-MP as an analysis tool rather than analyze the various weather data products directly. This indirect method was chosen for two reasons. First, the JPADS-MP must perform an internal analysis in order to generate navigational outputs; and second by using the actual mission planning tool the operators use, the warfighter is assured of a result with immediate operational application.

Finally, all forecast data used for this thesis was collected for YPG and used the Penn State University/National Center for Atmospheric Research Mesoscale Model 5 (MM5) forecast model. This analysis will need to be reaccomplished when AFWA changes from MM5 to the Weather Research and Forecasting (WRF) model. Time limitations prevented attempting to gather data from areas other than YPG. It is therefore worthwhile to use caution in applying the results of this research to other locales before additional data can be reviewed. The methodology used in this thesis will allow for such additional analysis with ease. This is an advantage of using the JPADS-MP as an analysis tool.

II. Literature Review

Overview

The Joint Precision Air Drop System is intended to address several recognized capability gaps. It is a family of systems that includes, but is not limited to, the JPADS Fly-Away Kit and several candidate guided parachute systems. This chapter will begin with a brief discussion of the historical environment that led to JPADS. It will then progress to a description of the systems that comprise JPADS. This will be cursory as JPADS is well covered in other documents and is not the actual focus of this thesis. Attention will be given to aspects of the JPADS-MP which were of specific use in this thesis. The chapter will conclude with a review of the weather data types and formats used in this research.

Historical Background

On 16 January, 1784, an American living abroad in France penned a letter to a friend concerning a revolutionary technology he had recently observed. The technology in question was a balloon capable of lifting two men into the air. This American saw more than a mere curiosity in the balloon. In fact, he had an extraordinarily prescient vision of what would stem from the invention.

On that day he wrote:

...where is the prince who can afford so to cover his country with troops for its defense, as that ten thousand men descending from the clouds might not in many places do an infinite deal of mischief, before a force could be brought together to repel them?

The writer of this letter was Benjamin Franklin. On 10 September, 1944, more than 160 years later, a copy of this quote was kept on the desk of another American located in England. This American was Lieutenant General Lewis H. Brereton and on that day in September, he was responsible for planning Operation MARKET – the allied airborne invasion of Holland (11:122).

Operation MARKET-GARDEN was a combined airborne and land based invasion. The First Allied Airborne Army was to drop in Holland and hold key bridges along the route to and across the Rhine. This was Operation MARKET. The British XXX Corps armored unit would drive up a narrow corridor of advance, relieving the airborne units as it went, until it crossed the Rhine. This was Operation GARDEN. If successful, it could have brought about an early end to the war. However, this was not to be.

MARKET-GARDEN would require three major airdrops of troops over three days. Once on the ground, the airborne units would require airdrop resupply. While it would be difficult to identify any one element that led to the failure of MARKET-GARDEN as being decisive, the lack of precision airdrop capability is clearly significant. History records abysmal airdrop accuracy. British airborne troops "watched in despair as thirty-five Stirling bomber-cargo planes dropped supplies everywhere but on the [drop] zones. Of eighty-seven tons of ammunition, food and supplies destined for the men of Arnhem, only twelve tons reached the troops. The remainder, widely scattered to the southwest, fell among the Germans" (11:376). This was not the first or the last time that airdrop would inadvertently supply the enemy rather than the defenders.



Figure 1. C-47's performing low altitude airdrop during Operation MARKET-GARDEN. (Source: http://www.qmfound.com/airborne2.gif)

In April of 1972, the forces of North Vietnam launched their Easter Offensive. It was an effort to overrun South Vietnam in one stroke. The key to South Vietnam was the capitol in Saigon. North Vietnamese forces planned to launch from Cambodia and drive the 90 mile distance down Highway 13 to the capitol. On this highway, approximately 26 miles from the Cambodian border, sat the city of An Loc. It was here that a major battle would ensue that would lead to a two month long siege. By 1972, the majority of American ground units have been withdrawn. The Army of the Republic of South Vietnam (ARVN) had about 6,000 troops in An Loc to defend against more than 35,000 North Vietnamese forces (19). It would fall to American air power to sustain them.



Figure 2. Route of attack on An Loc and surrounding area. (Source: http://www.vnafmamn.com/Valiant_Anloc.html)

Airdrop crews flying in support of the forces on the ground at An Loc faced a lethal curtain of fire including .51 caliber, 37mm, and 57mm Anti Aircraft Artillery (AAA) (9). On 11 May, the first SA-7 Strela, Infrared guided, Man Portable Air Defense (MANPAD) weapon was fired in the vicinity of An Loc (19). Prior to this time, the only technique that afforded an adequate level of precision was the Low Altitude Parachute Extraction System (LAPES). But such tactics proved to be suicidal in face of the anti-air environment around An Loc. The 374th Tactical Airlift Wing had an operating detachment at Tan Son Nhut Air Base and was tasked with the airlift mission for An Loc (9). They developed revolutionary techniques for high altitude airdrop called Ground Radar Aerial Delivery System (GRADS) and Adverse Weather Aerial Delivery System (AWADS). These techniques allowed for improved accuracy for airdrop from above 12,000 feet (2). The 374th also developed new parachute methodology. They devised a method for airdrop using a smaller 26 foot diameter "ring-slot" high velocity parachute

than the standard 64 foot diameter G-12 parachute canopy. The ring-slot chute served not to decelerate the load, but to stabilize it at it fell. Careful packaging allowed most types of loads to survive the landing (9).



Figure 3. C-130 performing LAPES cargo drop during the siege of Khe Sahn in 1968. (Source: http://www.qmfound.com/khe_sanh1.jpg)



Figure 4. 26' Ring Slot High Velocity Parachutes in flight. (Source: http://www.pioneeraero.com/pop-ups/2-14-IMAGE1.htm)

These new techniques allowed the defenders of An Loc to hold out against the overwhelming odds they faced. The Easter Offensive failed and South Vietnam survived for another three years. For the airdrop forces involved, the final tally of losses were 15

aircrew casualties, numerous wounded aircrew members, 37 aircraft damaged, and the loss of 2 C-123 and 3 C-130E aircraft (19).

The techniques developed by the 374th in the support of An Loc would last far beyond that South East Asian battlefield; more than 30 years later, versions of them are still in use. Airdrop would continue to play a key in military operations all over the world. Operations JUST CAUSE, PROVIDE PROMISE, ALLIED FORCE, ENDURING FREEDOM and others would see airdrop being called on time and again. In the intervening years, the threats faced by the aircrews in Vietnam have only intensified. Reaching an adequate level of precision using conventional techniques now places aircraft and their crews at unacceptably high risk.

The airdrop mission has evolved beyond the 1970's era solution. Methods are needed to operate outside the Weapons Engagement Zone (WEZ) of MANPADS and AAA while reaching totally unprecedented levels of precision. It is additionally desirable that airdrop be able to operate at an offset from the desired PI. Such a capability will allow covert teams to be resupplied via airdrop without their position being highlighted by overflight of the drop aircraft. Fortunately, threats and requirements are not the only thing to have evolved since the 1970's.

Since the advent of the Global Positioning System (GPS) in the 1990's more and more military systems have come to rely upon the navigation technology. The Joint Direct Attack Munition (JDAM) revolutionized precision engagement and has virtually become a household word. It would not take long for the technology that made the JDAM possible to begin transiting to the airdrop world. The stage is finally set for the Joint Precision Airdrop System – JPADS.

Joint Precision Air Drop System Overview

In traditional airdrop, the aircrew must fly the aircraft to a specific point in the sky, known as the Computed Air Release Point (CARP). The CARP is calculated using variables such as payload weight, drop altitude, aircraft velocity vector, wind velocity vector, and location of the intended Point of Impact (PI). One CARP corresponds to one PI. Miss the CARP and you miss the PI. Of course, hitting the CARP does not guarantee that you will hit the PI, but it is the point of maximum likelihood given the quality of the data input into the calculations. This is where weather sensitivities become important to understand.

JPADS is intended to revolutionize how airdrop works. This is about more than bringing GPS precision to CARP calculations though. JPADS is a Family of Systems that allows for precision airdrop to one or more PI from medium to high altitude with the option of significant standoff range (i.e., without the need to fly directly over the PI as in traditional airdrop). These capabilities allow for significant operational flexibility. For example a single aircraft could, in a single airdrop pass, drop loads to different PIs. Alternately, one or more aircraft could drop loads from a broad Launch Acceptability Region (LAR) to hit a single PI from various launch points.

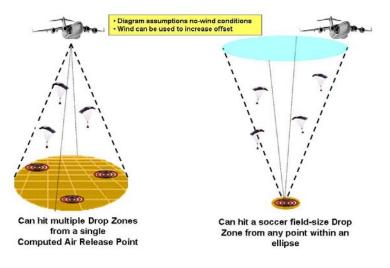


Figure 5. JPADS Guided Parachute drop capabilities (6)

These capabilities are important for tactical advantage as well as safety since JPADS allows aircrews to drop from altitudes and standoff ranges which are safe from enemy surface-to-air threats and terrain. And finally, the ability to drop on a PI without direct over-flight serves to further protect the aircraft and crew as well as to prevent highlighting the location of the PI and the airdrop recipient. Figure 6 shows the JPADS Systems Architecture Operational View (OV-1) Diagram. The OV-1 is a graphic depicting the high-level operational concept of the JPADS architecture.

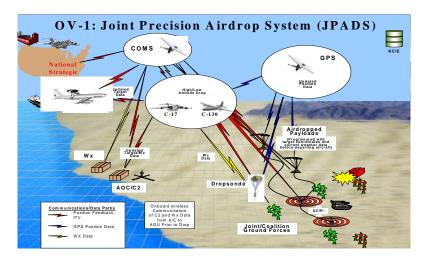


Figure 6. JPADS OV-1: Overall view of system activity. (6)

JPADS Physical Components.

JPADS consists of a roll-on/roll-off system suite for the aircraft, a mission planning element, and a variety of specialized Guided Parachute systems. The Air Force is the program lead for developing the aircraft systems which consists of the JPADS Fly-Away Kit, JPADS Mission Planner (JPADS-MP) software, and Global Positioning System (GPS) Dropsondes. The US Army has responsibility for the development of the Guided Parachute systems.

JPADS Fly-Away Kit.

The JPADS Fly-Away Kit is self-contained unit designed to give roll-on/roll-off JPADS system capability for an airlift platform such as the C-130 or C-17. The kit contains a Precision Air Drop System (PADS) software configured Panasonic CF-29 Toughbook (also known as the PADS Laptop Computer or PLC), a Global Positioning System Retransmission System (GPS – RTS), the Advanced PADS Interface Processor (APIP), and all necessary connections for the system and the aircraft. Figure 7 shows the JPADS Fly-Away Kit in its stowed and unstowed configuration. With its case, the Fly-Away Kit weighs 75 lbs. The Kit is developed by Planning Systems Inc, Draper Labs, and the Forecast Systems Lab of the NOAA (6).



Figure 7. JPADS Fly-Away Kit components. (1)

JPADS Guided Parachute Family.

Presently, several system types are under consideration. Among these are the Affordable Guided Airdrop System (AGAS), the Screamer, and the Sherpa. Each system differs in approach to the guided airdrop problem solution as well as in overall performance capabilities. This section will provide background on each system and how it fits into the JPADS architecture.

Affordable Guided Airdrop System (AGAS).

Developed in joint venture by Vertigo Inc and Capewell, AGAS is a family of systems for precision airdrop of loads from 200 to 10,000 pounds. It is intended to provide high accuracy and precision at low cost by utilizing off the shelf parachutes and rigging components and is essentially a strap-on guidance kit for the standard Container Delivery System (CDS). The AGAS system is compatible with existing inventory parachutes such as the G-12 and the 26' Ring Slot High Velocity Parachute. The heart of the system is the Autonomous Guidance Unit (AGU). The JPADS-MP generates a Wind Profile which is used to calculate a wind corrected flight trajectory. This trajectory is passed on to the AGAS AGU. The AGU monitors the actual flight path as compared to the nominal flight path. The flight path is then adjusted by "slipping" the parachute control risers. The Figure below displays the AGAS mission profile.

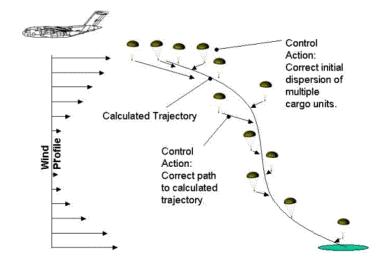


Figure 8. AGAS flight profile with Wind Profile corrections. (Source: http://www.vertigoinc.com/agas)

In testing, AGAS has proven to be highly accurate, and typically has the smallest Circular Error Probable (CEP) of the candidate systems. However, it also has the least horizontal standoff capability among the candidates at approximately 5 km. AGAS loads typically have a 14-15 minute Total Time Aloft. The following figure shows example AGAS miss distances and their associated CEPs.

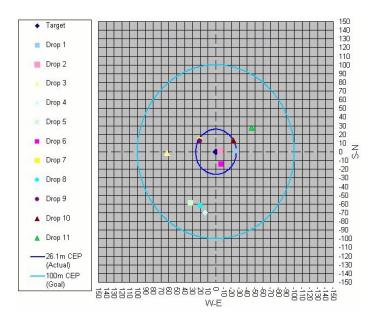


Figure 9. Sample drop score card and CEP achieved by the AGAS Guided Parachute system. (Source: http://www.vertigo-inc.com/agas/cep.jpg)

Sherpa.

The Sherpa guided parachute system is the product of Mist Mobility Integrated System Technology (MMIST). Sherpa is a family of four systems with load capacities ranging from 265 lbs to 2200 lbs. The Sherpa system uses a large Ram Air Parachute (RAP) which gives the system a significant glide range and maneuverability. The RAP affords Sherpa a horizontal standoff range of up to 20 km from a drop altitude of 25,000 feet. A unique feature of Sherpa is the option to provide terminal guidance via a hand control unit. Otherwise, the Sherpa uses an AGU to correct for windage errors with respect to the preplanned trajectory calculated by the JPADS-MP.



Figure 10. MMIST Sherpa prepares to land. (Source: http://www.mmist.ca/Sherpa.asp)

The following chart shows a series of Sherpa 2200 miss distances with associated 100, 200, and 300 m CEPs. Sherpa is a Commercial Off-The-Shelf (COTS) system already in use with US Marine Corps.

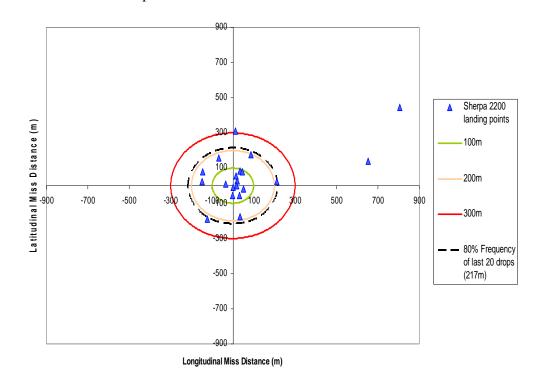


Figure 11. MMIST Sherpa sample drop score card and CEP. (Stoker, 2006)

Screamer.

Strong Enterprises with Robotek Engineering have developed the Screamer Precision Cargo Delivery System. The Screamer is unique in that it uses an undersized Ram Air Drogue (RAD) rather than a full size canopy. The RAD serves to stabilize and decelerate the payload as well provide steering capability. The use of the RAD also allows for a rapid decent from altitude and improved resistance to wind effects. However, due to its small size, the Screamer RAD is incapable of slowing the payload down for landing. This is accomplished by the deployment of one or more standard round, unguided parachutes (typically one or more G-11 parachutes) once the payload nears the surface. Once the recovery chute is deployed, the Screamer is considered to be ballistic, which is to say, at the mercy of the low-altitude winds. The figures below show the phases of Screamer flight, first under the RAD and then moments after the deployment of the Recovery Parachutes.

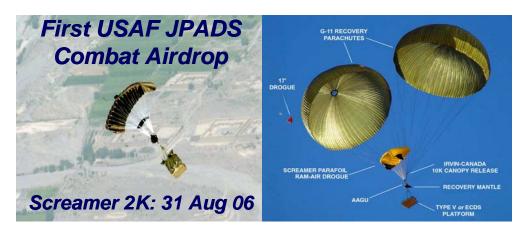


Figure 12. Phases of Screamer Flight. (17, 10) As with the other systems, Screamer navigation is accomplished via an AGU which

compares its real-time position with a preplanned trajectory using onboard GPS. The

Screamer has a glide ratio of 2.6:1. From a drop altitude of 25,000 feet, it has a standoff distance of about 7.2 miles and has a Total Time Aloft of approximately 8.6 minutes. The following Figure show example miss distances for the Screamer system and its associated CEPs.

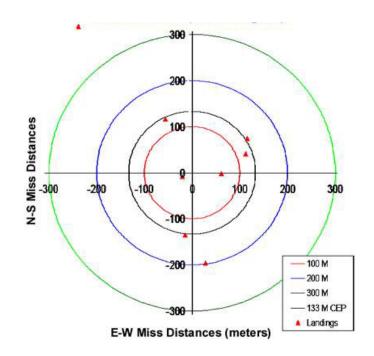


Figure 13. Scream drop score card and CEP. (10)

Joint Precision Airdrop System – Mission Planner (JPADS-MP).

The JPADS-MP is a combination of the PLC and the PADS software. Its ultimate use is to ensure that the cargo arrives at the desired PI. Figure 14 shows the functional design of JPADS as well as the data flow structure.

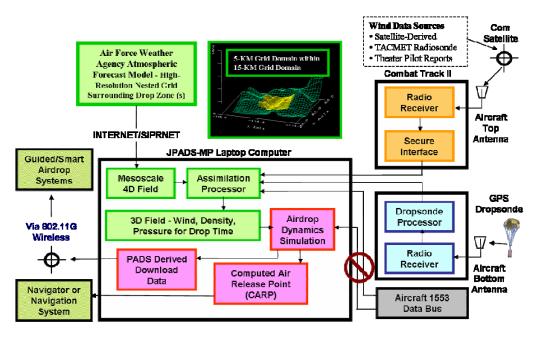


Figure 14. JPADS-MP system layout. (1)

The box at the center of the Figure shows the functions performed by the JPADS-MP. As can be seen, data flows into the JPADS-MP from the GPS dropsonde, Combat Track II messages, NIPRNET/SIPRNET, and from user input. Data flows out of the JPADS-MP to the user and to the Guided Airdrop Systems. A more comprehensive picture comes from the JPADS SV-4 System Functionality Description diagram in Figure 15.

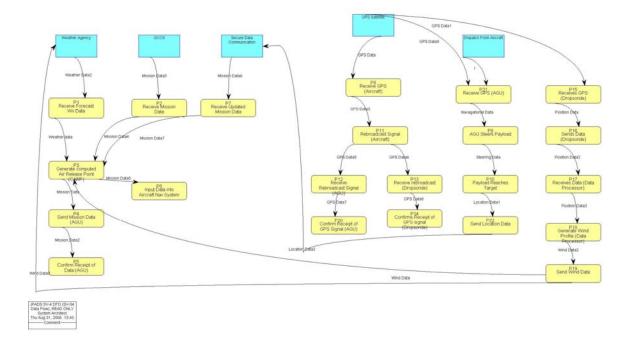


Figure 15. JPADS SV-4, Systems Functionality Description. (6)

These two diagrams (Figures 14 and 15) are important to providing the answer to the first Investigative Question: How does weather affect JPADS? (i.e., how does the JPADS-MP ingest and use weather data? What are the outputs?) The next diagram is a simplified version of the SV-4. All non-weather related items and data flows have been removed. This Systems Engineering product answers where weather enters the system and how that data flows within the system accomplish its functions.

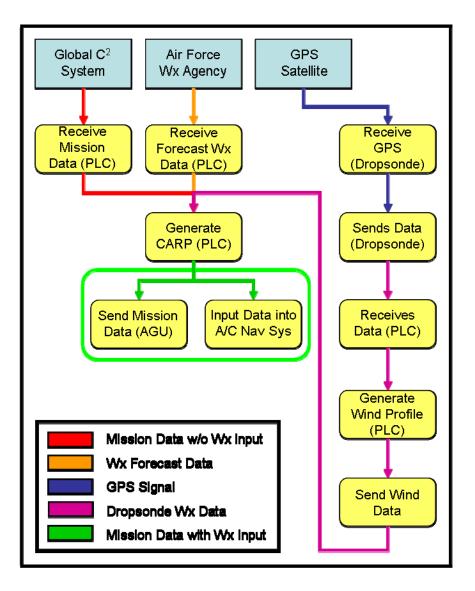


Figure 16. Weather Data Flows within JPADS.

In this color coded, streamlined version of the SV-4, the data flow is easy to follow. Looking at the functions at the bottom of the flow, we find (as one would expect) the AGU and the aircraft navigation system. What stands out as the critical function lies one step above these: Generate CARP – the Computed Air Release Point, a function of the PLC. The flows shown here are the primary ones available to operational users. There are options not shown here. These include Pilot Reports (PIREPS), Ballistic Winds, and Climatology. However, it turns out that the critical path is the same in all cases. The PADS Laptop Computer (PLC) requires at least two inputs to calculate the CARP: mission data and at least one type of weather data. The next sections will discus these inputs as well as the CARP in more detail.

Air Force Weather Agency (AFWA) Weather Forecasts.

The JPADS-MP uses 4 Dimensional forecast models in order to generate the best CARP solutions. These forecasts are called 4 Dimensional as they include x, y, and z spatial coordinates as well as a temporal coordinate. AFWA generates forecasts for YPG in three levels of resolution: 5km, 15km, and 45km. In 4 Dimensional forecast models, resolution refers to how closely spaced the data weather data points are on the x/y grid plane. Thus, 5km spacing is high resolution, containing much more data than a resolution with 45km spacing of grid points. Obviously, higher resolution means a larger data file, and thus greater band-width for transmission and longer download times.

The 5 and 15km models are run every 12 hours and the 45km model is run every 6 hours. One model run simulates 24 hours of weather. To get a forecast for a given time, say 1200Z, several options are available. The 5 and 15km forecasts are initiated at 0600Z and 1800Z. To get a forecast valid for 1200Z, you could use the model initiated on the current day at 0600 and take the predicted weather conditions 6 hours after model start. An 0600Z start time plus 6 hours lead time equals a valid time of 1200Z. Alternately, the 1800Z start time from the previous day with an 18 hour lead time also results in the desired 1200Z valid time on the current day. If the 45km resolution is

considered, five more forecasts (three from the current day and two from the previous) are available to predict the conditions at 1200Z.

A unique aspect of weather forecasting is that models are initiated "dry." This means that data such as humidity, dew point, and pressure are fed into them, but not information on precipitation or cloud effects. It is left to the model's weather physics to generate this information. This results in a certain amount of spin-up time being required by the model before it begins to provide realistic forecast results. It is this feature which calls into question how model lead time affects forecast accuracy.

A collection of these forecasts as well as corresponding weather balloon soundings covering the JPADS ACTD test activity from 20 June 2005 to 5 December 2006 has been provided by AFWA Detachment 3 for the purposes of this research. A list of the weather balloons specifically used for this analysis is included in Appendix H. This data is used to determine which forecasts can be used for CARP generation.

Computed Air Release Point (CARP) and Launch Acceptability Region (LAR).

Certainly, one of the keys to precise airdrop is positioning the drop aircraft in the proper position in space with respect to the PI, taking into account the variables of aircraft velocity as well as the wind velocity at each altitude from the drop level down to the ground. This point in space is traditionally known as the Computed Air Release Point (CARP). One of the chief functions of the JPADS-MP is creating a highly accurate CARP. This is accomplished by taking into account aircraft type, altitude, heading, airspeed, position, and ramp-angle, as well as parachute type, load weight, et cetera. To these variables, a final key ingredient is added: the wind profile. While the payloads are

guided, they are unpowered and cannot regain kinetic energy once spent. This makes a good knowledge of the air mass they are to fly through critical to hitting the PI. This CARP is then input (by hand) into the drop aircraft navigation system. Although a precise formulation of the CARP is not as critical for Guided Parachute systems, JPADS-MP is also used to improve the accuracy of cheaper, unguided parachute systems such as the High Velocity Container Delivery System (HV-CDS). Thus, operators need the best possible weather estimate to ensure accurate airdrop.

Since the Guided Parachute systems have the energy to fly to the PI from a large area, the JPADS-MP calculates a Launch Acceptability Region (LAR) in addition to the CARP. The LAR is an elliptical region which represents the approximate area in space from which a Guided Parachute system could successfully reach the designated PI given the weather inputs to the JPADS-MP. Mathematically, it is the solution space containing all feasible CARPS for the Guided Parachute systems for a given set of PI coordinates and weather inputs. It is important to note that this region is an approximation intended to give aircrews a good idea of the system limitations. The edge of the LAR should not be considered a precision measurement for drop purposes. To deal with this and other uncontrollable variables (such as the true weather vs. forecast) a safety factor of 11% is subtracted from the LAR ellipse. The safety factor is a user definable option within the JPADS-MP.

Figure 17 shows a comparison of airdrop missions using traditional, non-JPADS planning, as well as JPADS planned guided and unguided drops. Note that for unguided drops, only one CARP is available to hit the PI; while for guided drops an elliptical area

defined by an Early, Nominal, and Late CARP (the Launch Acceptability Region) is sufficient to hit the PI.

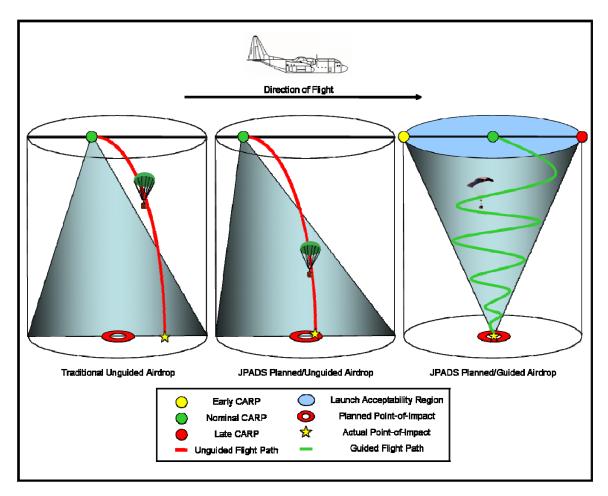


Figure 17. Comparison of Flight Profiles for traditional vs. guided airdrop options.

Screamer Recovery Chute Opening Point (OP).

In addition to the CARP/LAR, a third calculation is made by the JPADS-MP in support of the Screamer Guided Parachute system. Unlike the AGAS and Sherpa, the Screamer requires an additional AGU command beyond the CARP/LAR computation. This is for the recovery chute Opening Point (OP), also called the *pickle-point*. The correct calculation of this point is critical to hitting the PI since the Screamer payload is no longer guided from there. Figure 18 demonstrates the particular case of the Screamer system and its sensitivity to correct weather forecasting. The conic section indicates the volume of space in which the Screamer system has sufficient energy to maneuver to reach the Opening Point (OP) calculated by the JPADS-MP. Once the OP is reached, the Screamer payload is carried to the ground by the low-level winds. As Figure 18 shows, if these winds are correctly forecast, Screamer can hit the PI with high accuracy.

Conversely, poor forecasting leads to missing the PI.

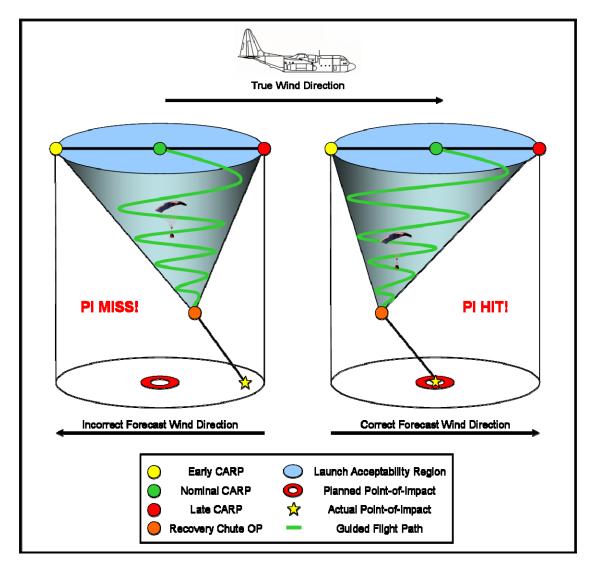


Figure 18. Screamer Flight Profile and weather sensitivities.

GPS Dropsonde.

This is a GPS instrumented unit that falls under parachute at a known velocity (typically 70 fps). The dropsonde is released from an aircraft to gather a sounding of the true weather in close geographical and temporal proximity to the planned airdrop location and time. The weather data gathered from the dropsonde can be integrated in flight with the preflight mission planning forecast to improve the preflight planned CARP/LAR and OP.

A natural assumption is that updating the CARP/LAR and OP generated from premission forecasts with sampled atmospheric data should improve their estimates. There are, however, potential faults in this assumption. First, dropsondes take time – both to fall and for their data to be assimilated into the model. In order to ensure adequate time, combat tactics call for dropsondes to be employed no later than ten minutes prior to the planned airdrop. Since airdrop missions are typically flown at approximately 150 mph, this equates to a minimum difference of more than 25 miles between where the atmosphere was sampled and where the actual drop will occur. It is easily conceivable that a dropsonde could be sampling weather on one side of a mountain ridge and the airdrop take place on the other side in completely different weather conditions. Additionally, in tests, dropsonde data reception becomes unreliable at low altitude – precisely the time when accurate information is most critical. Finally, there is the afore mentioned 11% factor of safety. This margin of safety is essentially energy in the bank for the on board guidance system to use should it encounter unexpected weather during descent. The question then becomes, is a dropsonde likely to ever dictate moving the LAR more than 11%, particularly considering the other limitations of dropsonde employment? Since dropsondes are a consumable, they add to mission cost as well as complexity.

Initial planning for this study called for an analysis of these questions. However, this was de-scoped from the thesis after consulting with Draper Labs concerning the LAR calculation performed by the JPADS-MP version used in this research. As the JPADS-MP continues to evolve, the calculation of LAR will change significantly rendering any work done valueless.

III. Methodology

Research Strategy

The first challenge with this research was in determining what was meant by weather sensitivity of the JPADS system and how to measure it. The intent was to apply statistical analysis to data, but to what data and how? The obvious answer to the first part of that question was the data recorded by AFWA Det 3 in support of the JPADS ACTD. For each test, a record was kept of the weather balloons launched in support of that day's missions as well as the associated valid weather forecasts. This data covers a period from 20 June 2005 to 5 December 2006. This was a lot of data, over 50 GB worth. The next question is how to analyze it. AFWA Det 3 has already begun looking at a direct, altitude by altitude, comparison between weather balloons and weather forecasts. This research compared the forecast wind velocity (heading and speed) against observed wind velocity for the three different forecast resolutions and various lead times. It was this initial research that prompted this thesis.

The JPADS-MP served as a means for providing an *apples-to-apples* analysis of the weather input options within the context of the Mission Planner, something which would be very useful to the user community. This is possible since the JPADS-MP can perform its navigational computations from either the forecasts alone or the weather balloons alone. This capability allows for the creation of Northing vs. Easting error comparisons, much like the Miss Distance charts for the various Guided Parachute systems shown previously in Chapter 2. To execute this, a standard mission scenario is used for evaluating all input data. The scenario is detailed as follows:

Mission Name: N (for due North Run-In heading) Drop Aircraft: C-130 Run-In: 360° Magnetic Weather Reference Point: YPG Site 16 (Weather Balloon Launch Location) Lat: N 33 19.800 Lon: W 114 19.800 Elevation: 1421 ft MSL Drop Altitude: 17500 ft AGL Airspeed: 135 KIAS Magnetic Variation: 12.346 W (deg) Total Ramp Load(s): 1 Loads To Drop This Pass: 1 Exit Location: RAMP Stick Type: Single Aircraft Altimeter Setting: 29.92 inches Hg Chute/System Type: Screamer Total Rigged (All-up) Weight (lbs): 8000 Flight Station (load c.g.): 677 Stick Position: Left Glide Safety Factor: 0.89 PI: YPG JPADS Center PI PI Coordinates: N 33 19.612/W 114 22.226 PI Elevation: 1249 ft MSL Ballistic Chute Type: 2 G11 Steerable Chute Type: 850 Sq-Ft (Screamer 10k System)

This *N* mission was used to generate CARP and OP navigation data from the historical weather data. These were then grouped by resolution for analysis. Analysis was performed in Excel, Matlab, and JMP 6. There are two stages to the analysis; the first compares the three resolutions, and the second compares lead time. The comparison variables are the population mean and variance. To ensure that the N mission was not introducing error, a second mission was tested on the 5 km data set. This *S* mission differed only in the Run-In heading of 180° Magnetic. The results indicated virtually no difference from the CARPs calculated in the N mission. The remainder of this chapter will detail how the N Mission was entered into the JPADS-MP and how the resulting data was captured and evaluated.

JPADS-MP Operation – N Mission

The JPADS-MP is developed by Draper Labs and Planning Systems Inc (PSI), and a complete user's manual is available from them. This discussion will be limited to the aspects of the JPADS-MP that were used in the execution of this research. Appendix G contains a sequence of figures that will provide the reader with sufficient familiarity with the JPADS-MP Graphical User Interface (GUI) to recreate the steps taken in this research. Upon starting the JPADS-MP, the user is presented with the main GUI page as shown in following figure:

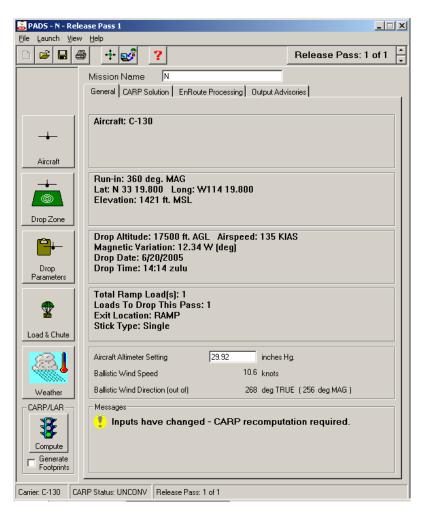


Figure 19. JPADS-MP main GUI.

For this study, the coordinates for Site 16 at YPG are used for the weather forecast reference point as this is the location from which the weather balloons were launched. This representative mission was created using an actual test point from the ACTD program, the only modification being a change in the Run-In heading to a cardinal direction. As a result, the PI is set as being the JPADS Center PI target at YPG, as used in testing. This is located 3.7 km from Site 16. It may have been better for the purposes of this analysis to have set Site 16 to be the PI. Unfortunately, this was realized too late for implementation. However, any error incurred by this is believed to be minimal when considering that the highest weather resolution was 5 km.

eatherAcquisition <u>W</u> ii	ndFileProductio	n <u>T</u> ools <u>H</u> elp		
rop Information			Wind File Production	
Planned Drop			Build Wind File	
Release Pass:	1 of 1			Build Status
Point of Impact	N 33 19. W 114 1		Best Available:	Idle
Drop Time : (zulu)	06/20/2 14:14	005	(Balloon only)	Cancel
Drop Altitude : (ft MSL)	18921			
/eather Acquisition				
Weather Source	Inventory	Age (hh:mm)-		
Dropsondes	0			
Pilot Report	0			
				(

Figure 20. Weather GUI

The *Weather GUI*, shown in Figure 20, is where most of the work in this research was done. The next step is to acquire weather data. This is done by selecting one of the

options under the Weather Acquisition section. The options relevant to this research are: Dropsondes, 4D Forecast, Balloon, and Climatology.

The JPADS-MP uses the 4D Forecasts generated by AFWA. These come in a format known as GRidded Information in Binary format (GRIB) files. Once these are downloaded, the *Browse* button is used to point the Weather Source GUI to the location of required GRIB files. Once the appropriate path is specified in the "GRIB Files Location" field, select the "Acquire Forecast" button. This will read the weather forecast into the JPADS-MP Environmental Data folder.

	ndFileProductio	n <u>T</u> ools <u>H</u> elp		
)rop Information			Wind File Production	
Planned Drop			Build Wind File	
Release Pass:	1 of 1			Build Status
Point of Impact	N 33 19. W 114 1		Best Available:	Idle
Drop Time : (zulu)	06/20/20 14:14	005	(LAPS Forecast-InSitu)	Cancel
Drop Altitude : (ft MSL)	18921			
Veather Acquisition				
Weather Source	Inventory	Age (hh:mm)-		
Dropsondes	0			
Pilot Report	0			

Figure 21. Weather GUI with 4D Forecast loaded.

In Figure 21, the 4D Forecast inventory now shows an increment of one and the Wind File Production section now has the options for wind file generation via LAPS Forecast. The Local Analysis and Prediction System (LAPS) is the most advanced modeling method included within the JPADS-MP. It allows complex modeling of wind interaction with terrain features such as how wind will flow over or around terrain obstacles. Select either the *Best Available* or *LAPS Forecast-only* (available under *Full Options*) to begin Wind File production.

) 🖻 日 🗧	ð 🕂 💕	?		Releas	e Pass: 1 of 1
	Mission Nar	ne N			
	General CAP	IP Solution EnR	oute Processing Output	Advisories	
	CARP	Early	Nominal	Late	
	Latitude	N 33 15.325	N 3318.490	N 33 21.654	
→	Longitude	W114 23.285	W114 22.457	W114 21.628	
	Short/Long	Short 8880	Short 2308	Long 4265	Yards
Aircraft	Left/Right	Right 104	Right 104	Right 104	Yards
<u> </u>	Altitude				
	TRUE (MSL)		18921	ft.	
Drop Zone	Pressure		17816	ft.	
DiopZone	Indicated		17816	ft.	
	AGL		17500	ft.	
	D-Value		-1105	ft.	
Drop Parameters	CARP Info				
	Convergence	Status	CONV		
	Optimization		Not Applicable		
X	95% CEP		N/A	Yards	
Load & Chute	Footprint Gen	eration Progress	0	200 Runs Max	
	- Load Spec	ific			
81	RAMP Load	1 • •			
Weather	Time of Fall F	rom Green Light (m	m:ss) 03:46		
CARP/LAR	- Messages-				
Compute		computatior ration reques	n completed succe ted.	essfully - No	footprint
Generate Footprints					

Figure 22. JPADS Main GUI CARP Solution TAB after successful CARP calculation.

Selecting *Compute* CARP will now automatically open the CARP solution tab. The CARP section shows the Latitude and Longitude of the Early, Nominal, and Late CARPs which also define the boundaries of the LAR. In order to collect this data, an Optical

Screen Reader tool was developed by Captain Ryan Eggert of the Air Force Research Laboratory Advanced Architecture and Integration Branch.

The Screen OCR tool reads the values in the Early, Nominal, and Late CARP coordinate boxes and copies them to a text file. In doing so, it also converts them from a DDD MM.mmm format to a DDD.dddddd format. The conversion to decimal degrees allows for easier mathematical operations later. Additionally, the Screen OCR copies the coordinates for the Screamer OP from its memory location and writes it to the same text file.

The method of building text files for analysis is to segregate the data by weather balloons. The Screen OCR allows for a new file to be opened and then to append subsequent data to this file. First, the CARP/LAR/OP is calculated for a weather balloon. This data is saved to a new file bearing the date and time of the balloon launch as the file name. Next, the CARP/LAR/OP is calculated for each weather forecast that was valid for the time of that weather balloon launch. Each new data set is appended to the text file resulting in a file similar to the one shown below:

📕 20050620 1414Z Dat	a.txt - Notepad						_ 🗆 ×
<u>File Edit Format View</u>							
33.255633 33.259800 33.259850 33.259850 33.258250 33.263250 33.263250 33.263250 33.258950 33.258950 33.255267	-114.387967 -114.377733 -114.380117 -114.377450 -114.380650 -114.379283 -114.379283 -114.380717	33.361117 33.365283 33.360383 33.360383 33.360383 33.360383 33.364433 33.364433 33.361900 33.360750	-114.360333 -114.350117 -114.352500 -114.349833 -114.352033 -114.352033 -114.352033 -114.351667 -114.351667 -114.35100	33.326595 33.327322 33.326797 33.327315 33.327645 33.327645 33.327645 33.327645 33.327645 33.3276815 33.326815	-114.371227 -114.370378 -114.370388 -114.370380 -114.370485 -114.370293 -114.370515 -114.370515 -114.370397	WX Balloon 15km Data 5km Data 5km Data 45km Data 45km Data 45km Data 45km Data 45km Data	14142 0600 ini 1800 ini 0600 ini 1800 ini 0600 ini 1200 ini 1800 ini
							•

Figure 23. Sample text file record of CARP and OP calculations from the JPADS-MP captured by the Screen OCR program.

As can be seen, each line represents a different weather input: weather balloon on the first line, followed by weather forecasts of varying resolution and initialization time. The coordinates of the CARPs and OP are to the left of the metadata. Capt Eggert also developed a CARP Analysis tool to generate Northing and Easting data from the raw coordinates captured by the Screen OCR.

The CARP analysis tool functions by comparing each weather forecast to the weather balloon data in line one of the text file. This results in a file similar to the one shown in Figure 24, below:

-								
-	0 1414Z NE Data.txt -	Notepad						
	⁼ <u>o</u> rmat ⊻iew <u>H</u> elp							
67 33 69 41 65 80 8	Early EW 953.587374 731.490987 979.956292 681.824147 774.933989 988.602194 809.192097 675.578024	-81.268824 467.752795 -81.272429 844.825326 367.936515 86.986301	OP NS OP EW 950.764528 729.028379 977.194776 679.421502 772.419161 806.543093 673.182890	Source Lead-T1 462.101579 -81.381192 467.649755 -844.728703 367.831778 86.876783 -40.680933	me 79.047517 78.116019 82.585575 69.085537 98.878861 86.961669 66.292295 77.278890	80.632102 22.404192 79.855759 20.518648 116.456310 70.650243 24.400489 30.500636	15km Data 15km Data 5km Data 45km Data 45km Data 45km Data 45km Data 45km Data	2014 814 2014 814 2014 1414 814 214 2014
◀								▼ ▶ //.

Figure 24. Sample text file containing output from the CARP Analysis Tool.

In this file, the data represents error in the forecasting. A value in the Nominal NS column of -40.578024 means that particular forecast generated a Nominal CARP coordinate that was 40.578024 m South of the correct Nominal CARP coordinate as defined by the Nominal CARP calculated from the weather balloon (an actual sampling of the atmosphere). Also note that, while the resolution data is unchanged, the initialization time has been replaced by the Lead-Time. This is accomplished by simply taking the difference between the weather balloon launch time and the forecast

initialization time. The data from each weather balloon (and its corresponding forecasts) is saved in a folder named for the day the balloons were launched on.

Once all the data has been run through the JPADS-MP and the final Northing/Easting data has been saved, the whole lot is read into Microsoft Excel. Excel is used to organize the data into continuous columns by resolution and then order them according to Lead-Time. The first order of business was to determine if a separate analysis would need to be performed on the Early, Nominal, and Late CARPS. However, comparing scatter diagrams for each type of CARP indicated this was unnecessary and that the Nominal CARP would suffice for all.

Each resolution is then entered into Matlab to test for Bivariate Normality. This test is taken from Walsh and Lynch's discussion on the Multivariate Normal Distribution (16:2). It was possible to code the test they describe into Matlab to produce a Goodness-of-Fit test for scaled distances to a Chi-Squared Distribution with n degrees of freedom. These are then fit to a regression model. The R²adjusted for the fit then give an indication of the GoF, where linearity correlates to normality. The Matlab input script and function are included in Appendix E and F, respectively.

Having passed this test, the data sets are then entered into JMP 6 for detailed analysis. JMP 6 was used to perform Analysis of Variance (ANOVA) as well as Multivariate Analysis. This was first performed for the full data set of each resolution in order to characterize each and determine if one was more favorable than the others in terms of mean (error) and variance. Then, each set was subdivided in order to examine the effect of Lead-Time on sample mean and variance. For 5 and 15 km data, there was insufficient data for anything other than a morning vs. afternoon comparison. The 45 km

data, however, was sufficient to group Lead-Times into seven bins of three hours each.

The Lead-Times for 45 km resolution range from approximately 2 to 23 hours.

IV. Results and Analysis

First Look

Before commencing the statistical analysis, the first objective was to verify that earlier assumptions made in setting up the test were valid. The following diagram is a scatter plot showing CARP data generated from the comparison of 5km weather data to their corresponding weather balloon derived CARP. The points on the graph show the error in the forecast based CARP with respect to the "true" weather balloon based CARP. The first check was to ensure that it would not be necessary to test the Early, Nominal, and Late CARPs individually, but rather, that one category would suffice for all. This chart shows that the errors for each type of CARP are perfectly correlated and validates the concept of analyzing only the Nominal CARP as a representative for the whole.

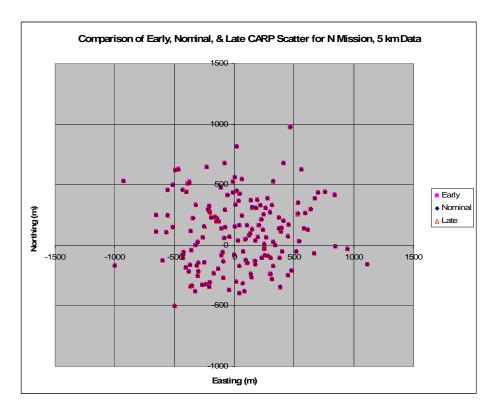


Figure 25. Scatter Plot of Northing and Easting Errors for Early, Nominal, and Late CARPs.

The next chart compares the N Mission used in this study with a notional S Mission. The only difference being a Run-In heading of 180° magnetic as opposed to 360° magnetic. The purpose of this test is to determine if the aircraft velocity vector played a significant role in the observed CARP errors. As can be seen below, there is excellent correlation between the N and S Missions, discounting any such concerns.

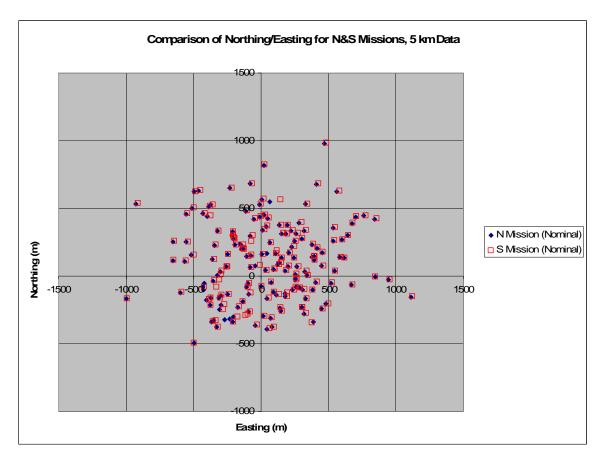


Figure 26. Scatter Plot of Northing and Easting errors of Nominal CARP comparing results from N and S Missions.

Figure 27 shows the results of the full data set. The upper chart shows the CARP errors in Northing and Easting between weather balloon and weather forecast inputs; the lower chart displays the same errors for the Screamer OPs.

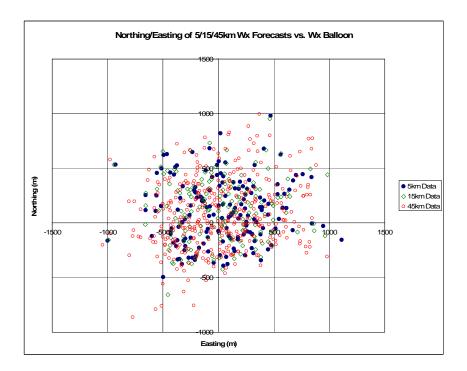


Figure 27. Scatter Plot of Northing and Easting errors of Nominal CARPs at 5, 15, and 45km Resolution

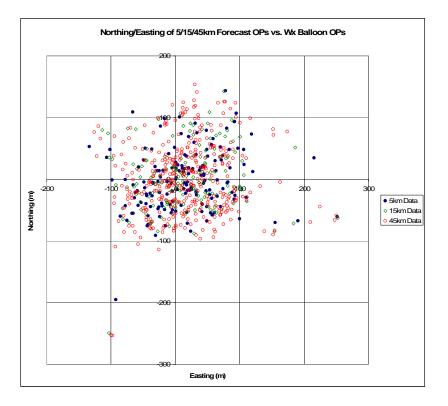


Figure 28. Scatter Plot of Northing and Easting errors of OPs at 5, 15, and 45km Resolutions

Goodness of Fit (GoF) Testing for Bivariate Normality

The next assumption to check is that of Bivariate Normal distribution of the data. As mentioned in Chapter 3, this is accomplished by fitting a line to a comparison of scaled distances to a Chi-Square distribution. As can be seen in the figure below, The CARP error data is a good fit to Bivariate Normal. However, the OP data is strongly influenced by outliers which, when included in the line-fit calculation, cause the OP data to fail the GoF test. Exclusion of these outliers allows for fits (shown in green on the charts) with R2 adjusted in the range of 0.98 - 0.99; clearly an excellent fit. Unfortunately, using historical data, there is no way to account for the cause of these outliers. Therefore, for the purposes of this study, they will not be removed.

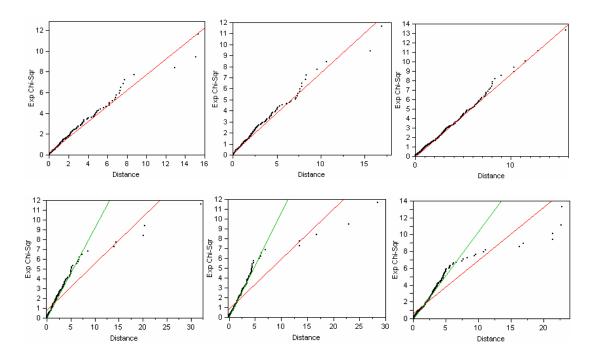


Figure 29. Bivariate Normal Goodness-of-Fit test applied to CARP data (top row) and OP data (bottom row) at 5, 15, and 45km Resolution (columns 1,2, and 3 respectively).

Since this establishes Bivariate Normal as a good distribution to describe the data, we now move on to analyzing the data in that light. The next series of figures will display statistical data necessary to answer whether there is an ideal weather forecast resolution for calculating the CARP.

Figure 30 shows the CARP error scatter for the 5, 15, and 45km resolution data. The green lines indicate the mean value for Northing and Easting. The solid red line is the Least Squares regression fit and the broken red lines indicate the 95% confidence interval around the fit. The fit is indicative of the correlation between Northing and Easting. The aqua line and shaded region is the 95% density ellipse for the data set. It is worth noting that in both the 5 and 15km resolutions, H₀ (there is no correlation between Northing and Easting) cannot be rejected at $\alpha = 0.05$. However, for the 45km resolution, H₀ is rejected at $\alpha = 0.05$. This can been seen in Figure 30, as the 95% confidence interval for the 45km data does not include a line of zero slope.

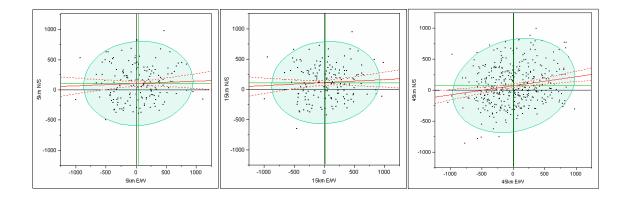


Figure 30. Scatter Plot of Northing and Easting errors with mean errors (green lines), correlation (solid red line), 95% confidence interval on correlation (dotted red line), and 95% density ellipse displayed for 5, 15, and 45km Resolutions.

This positive correlation was unexpected. As with the question of outlier data in the OP analysis, there is no clear cut answer to the source of this correlation. It does call into question the N Mission setup as a possible explanation. Weather data (both balloon and forecast) records wind direction using true headings. However, aircrews typically plan using magnetic headings. Since the N Mission borrowed its details from an actual mission, aircraft Run-In headings were entered using magnetic headings. The magnetic variance at YPG is approximately 13°. It is unknown if this plays a role in the observed correlation or not as there was insufficient time for testing after the discovery of the anomaly.

What is clear from these figures is the effect of resolution on both the means and variance of CARP errors. General improvement in mean error is seen as resolution decreases from 5km to 45km. However, the finer resolutions (i.e., 5 and 15km) have lower variance than does the 45km resolution. Additionally, all three resolutions exhibit a marked Northing error. The following tables provide summary statistics for Figure 30. The sample means and standard deviations are contained in the Correlation Table. Complete output from JMP 6 is included for all data in Appendix B, C, and D.

Linear Fit

5km N/S = 104.20727 + 0.0403265 5km E/W 15km N/S = 111.32849 + 0.0493597 15km E/W 45km N/S = 72.951882 + 0.1476738 45km E/W

Summary of Fit

RSquare	0.002742
RSquare Adj	-0.00349
RSquare	0.004211
RSquare Adj	-0.00197
RSquare	0.035599
RSquare Adj	0.033014

Analysis of Variance

Source	DF	Sum of Squares	Mean Square	F Ratio
Model	1	36162	36162.2	0.4400
Error	160	13149793	82186.2	Prob > F
C. Total	161	13185955		0.5081
Source	DF	Sum of Squares	Mean Square	F Ratio
Model	1	53276	53275.6	0.6808
Error	161	12599340	78256.8	Prob > F
C. Total	162	12652615		0.4105
Source	DF	Sum of Squares	Mean Square	F Ratio
Model	1	1295163	1295163	13.7687
Error	373	35086535	94066	Prob > F
C. Total	374	36381698		0.0002

Correlation

Variable	Mean	Std Dev	Correlation	Signif. Prob	Number
5km E/W	36.49546	371.6411	0.052369	0.5081	162
5km N/S	105.679	286.1824			
Variable	Mean	Std Dev	Correlation	Signif. Prob	Number
15km E/W	22.48796	367.396	0.064889	0.4105	163
15km N/S	112.4385	279.4684			
Variable	Mean	Std Dev	Correlation	Signif. Prob	Number
45km E/W	7.961625	398.495	0.188678	0.0002	375
45km N/S	74.12761	311.893			

Figure 31. JMP 6 Statistical Output for 5, 15, and 45km Resolution CARP data.

The answer to the question of resolution appears to be that that 45km data provides the lowest mean error, but the greatest variance. The next question is that of the effect of Lead-Time on CARP error. Figures 32 and 33 show the frequency of Lead-Times for each data set. Recall that Lead-Time is the delta between forecast initialization and the planned drop time. Since the historical data used was never intended for this type of study, it presents certain difficulties which will now be addressed. Testing at YPG typically occurs in two temporal groups: before noon and after noon local time. Since the 5 and 15km forecasts are only generated twice per day (at 0600Z and 1800Z), there are significantly less data points available for the Lead-Time study at these resolutions than for the 45km data (which is generated every 6 hours). Ideally, there would be at least thirty data points for each hour of Lead-Team to allow for a complete comparison; unfortunately that is not the case. In order to ensure enough data for statistical significance, Lead-Times must be grouped together in "bins." Due to the general paucity of data at the 5 and 15km resolution, these were lumped into two bins at the natural break point in the histogram. This compares Short Lead-Times (8 to 17 hours) to Long Lead-Times (17-24 hours).

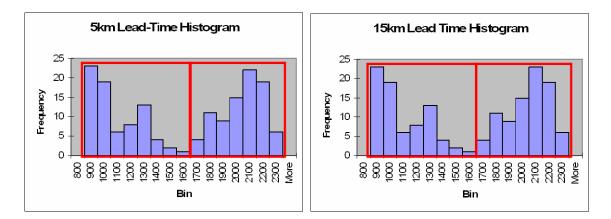


Figure 32. Lead-Time Histogram for 5 and 15km Resolution data.

The 45km data is more extensive, but still requires grouping for best results. In this case there are seven bins containing three hours of data each with a range from 2 hours of Lead-Time out to 23 hours – a much more complete set of observations.

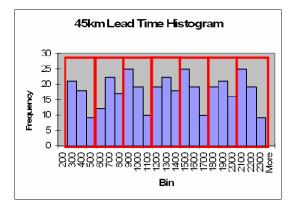


Figure 33. Lead-Time Histogram for 45km Resolution Data.

The analysis begins as before; this time with Northing and Easting error plots differentiated by Lead-Time as well as Resolution. We then move on to a One-Way Layout to further investigate the behavior of the means and variances as Lead-Time is adjusted.

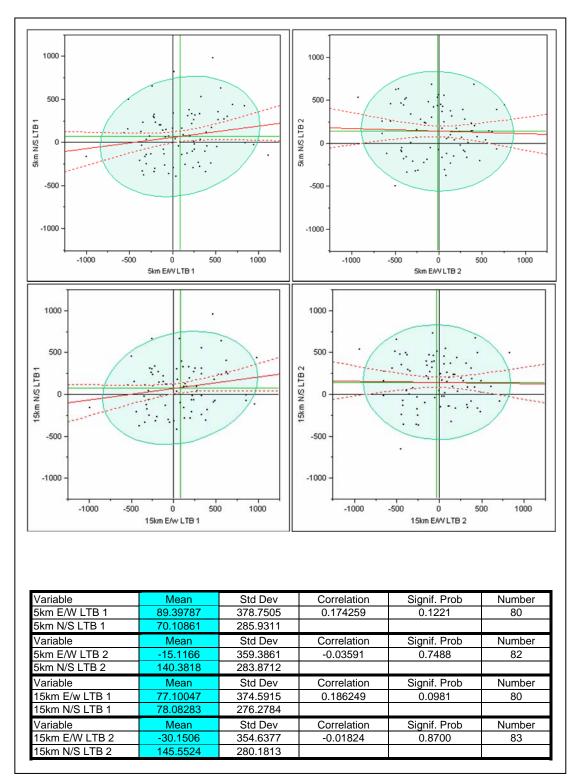


Figure 34. Scatter Plot of Northing and Easting errors for 5 and 15km Resolution sorted by Lead-Time bins as well as the associated statistical data.

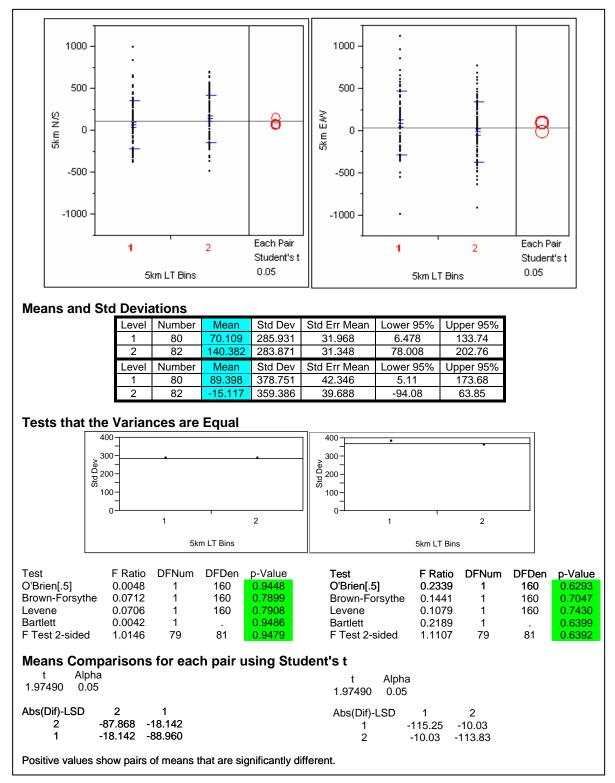


Figure 35. Results of JMP 6 Oneway Layout Analysis for 5km Resolution data. Northing data is on the left side of the figure and Easting is on the right for ease of comparison.

Figure 35 is a combination of the JMP 6 output for the 5km resolution forecast data. It allows for a side-by-side comparison of the means and variances of both the Northing and Easting data. The first two graphs show the distance errors with their associated bin number. The inner set of blue dashes indicates the mean of that set and its confidence interval. The outer set of blue dashes indicates 1 Standard Deviation. The red rings to the right of the chart are a visualization tool for comparing means. When this data is displayed in JMP 6, selecting one ring will cause it to be highlighted with a thick red ring (as opposed to a standard thin, black ring). Subsequently, all groups whose means are not significantly different change from black rings to red rings. Groups with significantly different means become gray. This test indicates that there is no significant difference in the means of Lead-Time bins 1 and 2 in either the Northing or the Easting data. The lower portion of the table is a test to verify that the variance between the bins is not significant. JMP 6 applies five different methods to this evaluation. In each case, the high p-Value indicates failure to reject H_0 : the variances between bins are equal. Figure 36 presents the same analysis for the 15km resolution data with comparable results.

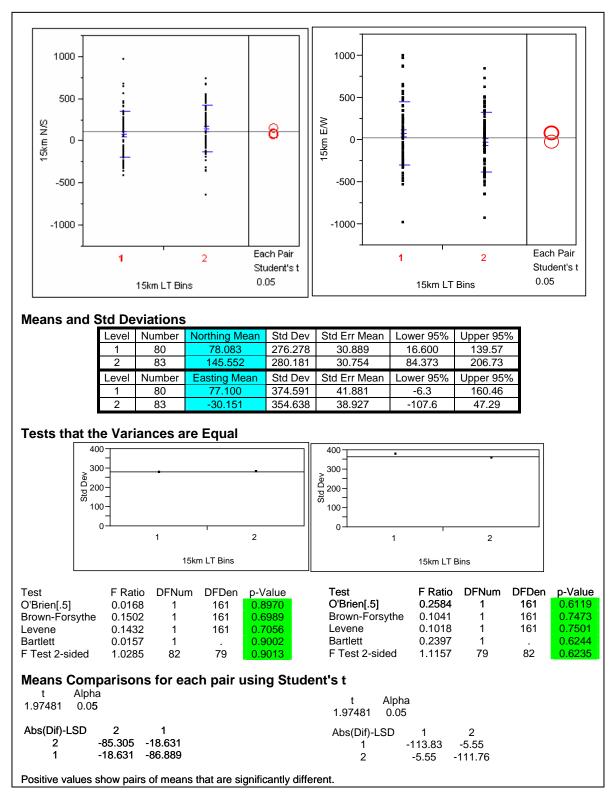


Figure 36. Results of JMP 6 Oneway Layout Analysis for 15km Resolution data. Northing data is on the left side of the figure and Easting is on the right for ease of comparison. Again, means and variances do not appear to vary significantly between Lead-Times at 15km resolution. We move on now to the 45km resolution data. Figures 37 and 38 display the CARP error data for each of the seven Lead-Time bins of the 45km data. In this sequence of charts, the Easting mean remains relatively close to zero with the greatest deviation occurring in Lead-Time bin 5, which represents data Lead-Time of 15 to 17 hours. Of more interest are the results of the Northing mean. For the first two bins (2 - 8 hours Lead-Time), the Northing mean is very close to zero. The precise values of the means are highlighted in Figure 39. The trend of the Northing error mean is generally worse beyond 8 hours of Lead-Time.

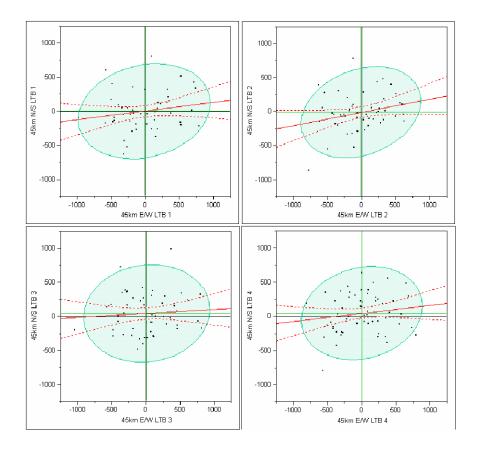


Figure 37. Scatter Plot of 45km Resolution Northing and Easting error sorted by bin, Bins 1 – 4.

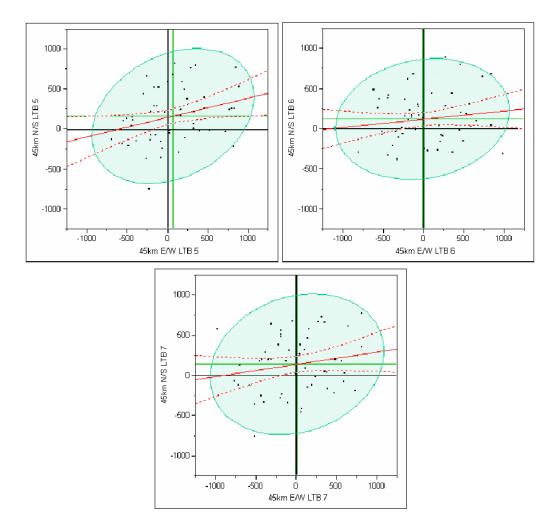


Figure 38. Scatter Plot of 45km Resolution Northing and Easting error sorted by bin, Bins 5 – 7.

Bin 5 is again the location of largest mean error, this time for the Northing error. Additionally, Bin 5 shows the greatest level of potential correlation between Northing and Easting error.

The next series of tables presents the Oneway Layout for the 45km resolution forecast data. There are some differences in the data presented here due to the addition levels (i.e., bins) available for comparison. First is the Tukey-Kramer HSD (Honestly Significant Difference) Test in addition to the pair-wise Student's t Test.

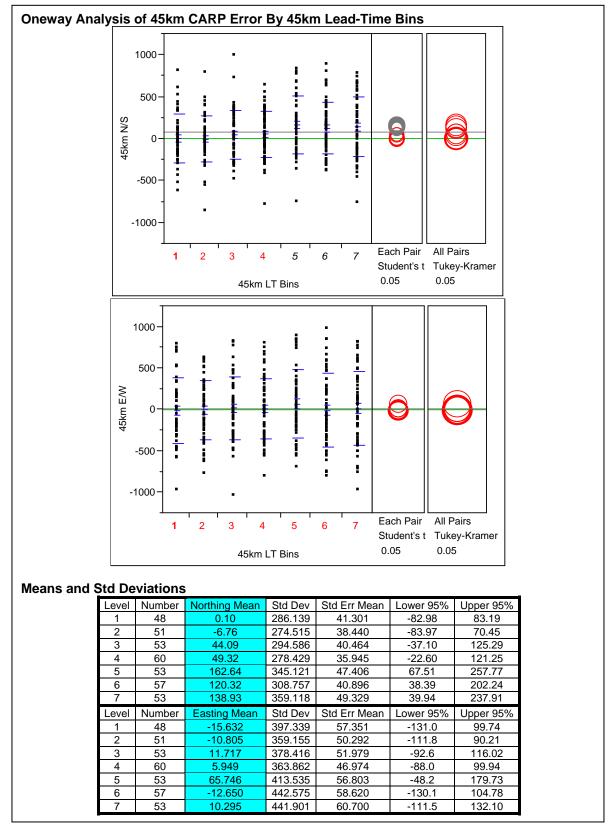


Figure 39. JMP 6 Oneway Layout Analysis of 45km Resolution data, Part 1.

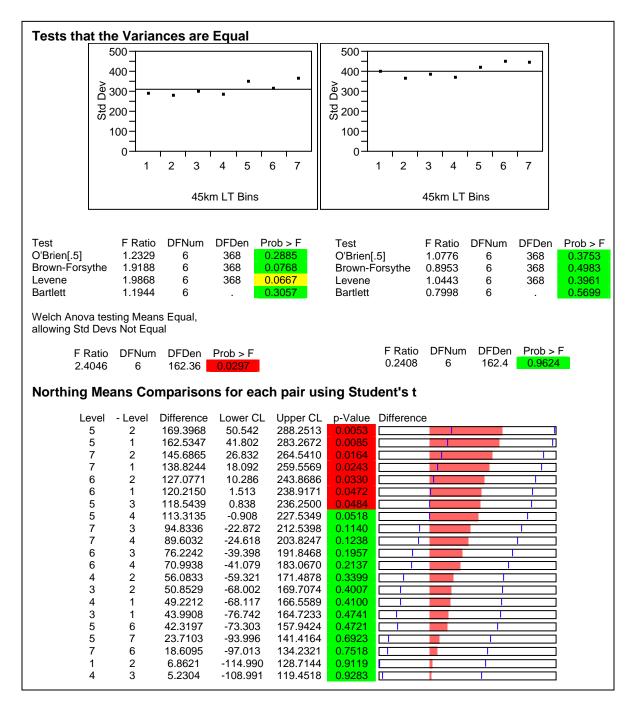


Figure 40. JMP 6 Oneway Layout Analysis of 45km Resolution data, Part 2

As previously mentioned, the rings to the right of the data plots are a visual tool for comparing the means of each group Based on the Student's t Test, we see two groups of similar means: 1,2,3, and 4 in the first group and 5,6, and 7 in the second. A green line has been added to the plot to indicate zero mean error. The black line on the plot displays the overall mean error. As can be seen, the first grouping of means fall below the black line while the second grouping lies above it. This corresponds (as expected) with the scatter plot data indicating generally greater error as Lead-Time increases, with the greatest error located in bin 5. Note the data table for the Student's t comparisons. Cells highlighted in red indicate a p-Value less than $\alpha = 0.05$ and thus a rejection of H₀: no significant difference in means between bins.

The Tukey-Kramer HSD Test is used here in addition to the Student's t because the number of observations in each bin is unequal. In such conditions, Tukey-Kramer is intended to provide a conservative test for difference in the means. In this case, the Tukey-Kramer result differs from the Student's t, indicating that that H_0 cannot be rejected under that test. For the Easting error data, both Student's t and Tukey-Kramer fail to reject H_0 , which is wholly expected from a visual inspection of the plot.

In both cases, bins with shorter Lead-Times display smaller Standard Deviations, and hence, smaller variance. However, the testing suite for equal variance fails to reject H_0 : no significant difference in variances between bins. The complete JMP 6 report is available in Appendices B, C, and D.

Next we will briefly examine the Screamer Opening Point (OP) error scatter plots. The improvement in overall error as compared to the CARP calculations is immediately apparent. These diagrams are on a 300x300m plot as opposed to the 1250x1250m plots for CARP errors. The radius for 95% density ellipse radius is about 150m at all resolutions. Additionally, it is not possible to reject H_0 (no correlation between Northing and Easting) for the OPs at any resolution. The mean Northing and Easting errors remain

virtually constant across all three resolutions, as can be seen in Figure 41. It is worth recalling that this data includes anomalous outliers. These were left in the data set as there is no method available for determining their cause (and thus justify their removal) from the historical data.

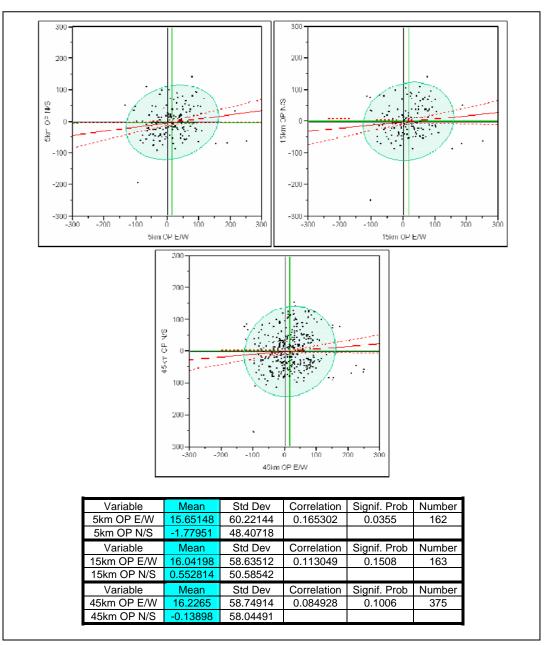


Figure 41. Scatter Plot of Northing and Easting errors for 5, 15, and 45km Resolution OP data as well as the associated statistical data.

This chapter concludes with a quick look at Yuma Proving Ground Site 16, the location where this weather data was collected. Figure 42 shows the locations of YPG Site 16 as well as the JPADS Center PI drop zone.

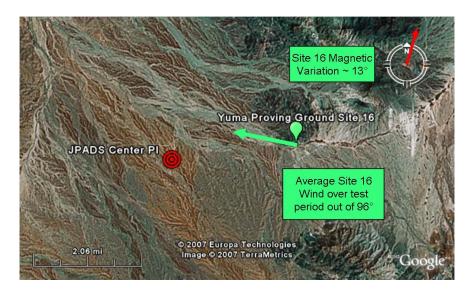


Figure 42. Satellite view of Yuma Proving Ground Site 16 and JPADS Center PI. (Source: Google Earth)

As can be seen in the figure, the average wind at Site 16 follows the local terrain.

Wind in this area can generally be expected to flow up the valley during daylight heating and down the valley during night-time cooling. The effect is shown in the diagram below:

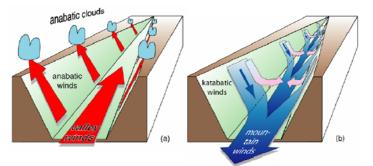


Figure 43. Wind flow effects due to terrain and diurnal heating effects. (15:406)

It is reasonable to believe that this local weather effect may be responsible for the greater variability observed in CARP Easting errors. However, certainty would require testing at other locations. Finally, Figure 44 is a composite image showing a Bivarite Normal distribution generated in Matlab from the 45km data with an overlay of the Eigen vectors of the distribution. The Eigen vectors are collinear with the axes of the ellipse. The angle between the Eigen vectors and the y-axis is approximately 18.7°. While this value is greater than the magnetic variance for the area, it is not by much. Also, it is aligned very closely to the direction of True North, as can been in Figure 44 below:

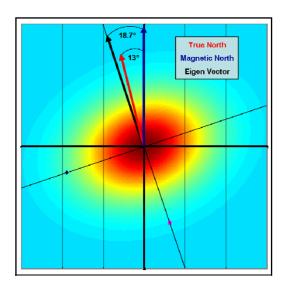


Figure 44. Overlay of Eigen Vectors on 45km Resolution derived Bivariate Normal distribution along with the approximate relationship to True and Magnetic North.

This argues in favor of the idea that inputting magnetic heading rather than true may be

the cause of the apparent correlation.

V. Summary and Conclusions

The Bottom Line

This study set out to determine how JPADS-MP outputs were sensitive to weather inputs. The main question being: are there ideal weather inputs to obtain the most accurate outputs? To answer this question required gaining a detailed understanding of how these three entities (input-system-output) interact. Not surprisingly, some questions remain. However there are clear, useful results from this research:

- 45km resolution weather forecast data has lower mean CARP calculation error than does 5 or 15km resolution weather forecast data. (*More accurate*)
- 45km resolution weather forecast data has higher variance in CARP calculation error than does 5 or 15km resolution forecast data. (*Less precise*)
- 2 to 8 hours of Lead-Time offers lowest mean CARP calculation error with least variance.
- Screamer Opening Point CARP calculation errors do not appear to depend upon weather forecast resolution.
- For the JPADS-MP, there appears to be no advantage to using higher resolution weather forecast data. High resolution weather forecasts have larger file sizes requiring greater bandwidth utilization and more time to acquire. Further, they are only run twice daily as opposed to the four times daily run of the 45km resolution data. It is clear that within the context of the JPADS-MP *as-is*, the lower resolution 45km weather forecast data is of greatest value to the user.

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Future Research

This study actually had two goals: the first, to attempt to answer the stated research question; the second, to determine if further research on this particular topic was needed. The answer to the second question is clearly: yes! Now that there is a better understanding of the system, the best way to achieve definitive results is to conduct a Designed Experiment rather than attempting to make do with historical data. While this is a time intensive process, this study has produced sufficiently interesting results to warrant the effort. Goals for a follow-on study should include the following:

- Gather weather balloon and weather forecast data from several, geographically diverse locations.
- Gather a sufficient number of weather forecasts at each resolution to allow full coverage of potential Lead-Times as well as to avoid the need of grouping Lead-Times into bins (as well as an equal number of observations at each level).
- Determine if entering navigation data using magnetic heading versus true heading is a source of error.
- Attempt to determine cause of the excessive Northing errors.

The final recommendation for future research is to address deficiencies in the method for acquiring climatology data for JPADS. JPADS-MP uses climatology data as filler whenever data necessary for calculations is not available from the current weather input data. Additionally, climatology may be used as a last resort for mission planning should no other source of weather input be available. At present, climatology is acquired

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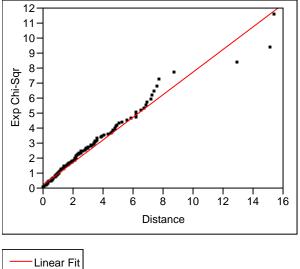
by sending a request to the Air Force Combat Climatology Center and waiting for them to provide the data. This can take as long as two weeks. AFIT's Engineering Physics department has developed an excellent software alternative known as the Laser Environmental Effects Definition and Reference (LEEDR), which could allow the JPADS-MP user to have the ability to generate accurate climatology data on the same PADS Laptop Computer that they use for the JPADS-MP. This would require only minor modifications to allow the two programs to talk to each other (certainly a Systems Engineering area of expertise!) and would provide immediate and tangible improvement in the system for the warfighter.

Appendix A: List of Acronyms

AAA	Anti-Aircraft Artillery
AGU	Airborne Guidance Unit
ACTD	Advanced Concept Technology Demonstration
AFIT	Air Force Institute of Tecnology
AFRL	Air Force Research Laboratory
AFWA	Air Force Weather Agency
AGAS	Affordable Guided Parachute System
AGL	Above Ground Level
ANOVA	Analysis of Variance
APIP	Advanced PADS Interface Processor
AWADS	Adverse Weather Aerial Delivery System
CARP	Computed Air Release Point
CDS	Container Delivery System
CEP	Circular Error Probable
DoD	Department of Defense
GoF	Goodness of Fit
GPS	Global Positioning System
GRP-RTS	GPS Re-Transmit Kit
GRADS	Ground Radar Air Drop System
GRIB	Gridded Information in Binary Format
GUI	Graphical User Interface
HVCDS	High Velocity Container Delivery System

JPADS	Joint Precision Air Drop System
JPADS-MP	JPADS-Mission Planner
JDAM	Joint Direct Attack Munition
LAPES	Low Altitude Parachute Extraction System
LAPS	Local Analysis and Prediction System
LAR	Launch Acceptability Region
LEEDR	Laser Environmental Effects Definition and Reference
MM5	Mesoscale Model 5
MMIST	Mist Mobility Integrated Systems Technology
NOAA	National Oceanic and Atmospheric Administration
OP	Opening Point
PADS	Precision Air Drop System
PI	Point of Impact
PIREP	Pilot Report
PLC	PADS Laptop Computer
PSI	Planning System, Inc.
RAD	Ram Air Drogue
WEZ	Weapons Engagement Zone
WRF	Weather Research and Forecasting
YPG	Yuma Proving Ground

Appendix B: 5km JMP 6 Analysis Output



Bivariate Fit of Exp Chi-Sqr By Distance (Nominal CARP)

Linear Fit

Exp Chi-Sqr = 0.2131093 + 0.7523327 Distance

Summary of Fit

RSquare	0.97758
RSquare Adj	0.97744
Root Mean Square Error	0.29726
Mean of Response	1.995723
Observations (or Sum Wgts)	162

Lack Of Fit

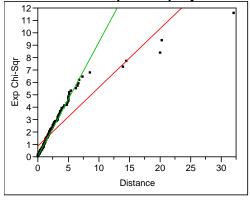
Source	DF	Sum of Squares	Mean Square	F Ratio
Lack Of Fit	158	14.137851	0.089480	655.7718
Pure Error	2	0.000273	0.000136	Prob > F
Total Error	160	14.138123		0.0015
				Max RSq
				1.0000

Analysis of	Variance		
Source	DF	Sum of Squares	Mean Square
Model	1	616.46791	616.468
Error	160	14.13812	0.088
C. Total	161	630.60603	
Parameter	Estimates		

Term	Estimate	Std Error	t Ratio	Prob> t
Intercept	0.2131093	0.031638	6.74	<.0001
Distance	0.7523327	0.009007	83.53	<.0001

F Ratio 6976.517 Prob > F <.0001

Bivariate Fit of Exp Chi-Sqr By Distance (Nominal OP)



Linear Fit

Linear Fit

Exp Chi-Sqr = 0.8372231 + 0.4746891 Distance

Summary of Fit

RSquare	0.820416
RSquare Adj	0.819294
Root Mean Square Error	0.841303
Mean of Response	1.995723
Observations (or Sum Wgts)	162

Analysis of Variance

Source	DF	Sum of Squares	Mean Square	F Ratio
Model	1	517.35942	517.359	730.9491
Error	160	113.24661	0.708	Prob > F
C. Total	161	630.60603		<.0001

Parameter Estimates

Term	Estimate	Std Error	t Ratio	Prob> t
Intercept	0.8372231	0.078773	10.63	<.0001
Distance	0.4746891	0.017558	27.04	<.0001

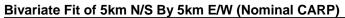
Linear Fit

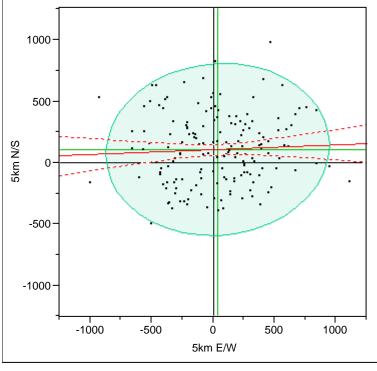
Exp Chi-Sqr = 0.060006 + 0.9201943 Distance

Summary of Fit

RSquare RSquare Adj Root Mean Squar Mean of Respons Observations (or S	e		0.995629 0.99560 0.1001 1.74638 156	1 1 7	
Analysis of V	ariance				
Source	DF	Sum	of Squares	Mean Sq	uare
Model	1		351.59272		.593
Error	154		1.54338	0	.010
C. Total	155		353.13610		
Parameter Estimates					
Term	Estima		Std Error	t Ratio	Prob> t
Intercept	0.06000		0.012054	4.98	<.0001
Distance	0.920194	13	0.004913	187.30	<.0001

F Ratio 35082.19 Prob > F<.0001





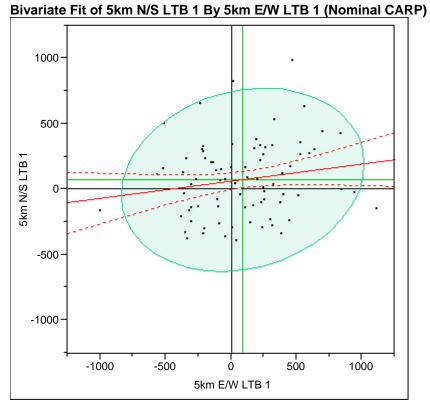
Linear Fit

5km N/S = 104.20727 + 0.0403265 5km E/W

Summary of Fit

Analysis of Variance			_
Observations (or Sum Wgts)		162	
Mean of Response		105.679	
Root Mean Square Error		286.6814	
RSquare Adj		-0.00349	
RSquare		0.002742	

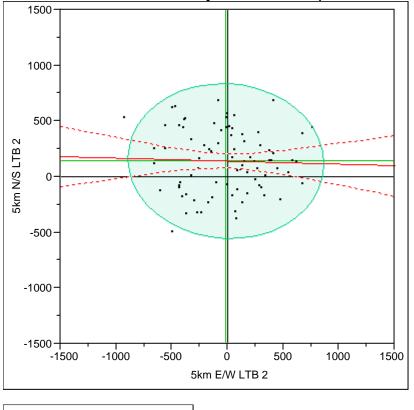
Source Model Error	DF 1 160	Sum of Squares 36162 13149793		luare 62.2 86.2	F Ratio 0.4400 Prob > F	
C. Total	161	13185955	01		0.5081	
Parameter Es	stimates					
Term	Estimate	Std Error	t Ratio	Prob> t		
Intercept	104.20727	22.63283	4.60	<.0001		
5km E/W	0.0403265	0.060794	0.66	0.5081		
Correlation						
Variable 5km E/W 5km N/S	Mean 36.49546 105.679	Std Dev 371.6411 286.1824	Correlation 0.052369	Signif. 0.	Prob 5081	Number 162



Linear Fit

5km N/S LTB 1 = 58.347947 + 0.1315542 5km E/W LTB 1

0.030366 0.017935 283.3554 70.10861 80						
Sum of S	quares N	lean Square	F Ratio			
19	6129.5	196129	2.4428			
626	2642.5	80290	Prob > F			
645	8772.0		0.1221			
Parameter Estimates						
Estimate	Std Error	t Ratio	Prob> t			
58.347947	32.56149	1.79	0.0770			
0.1315542	0.084171	1.56	0.1221			
Mean 89.39787 70.10861	Std Dev 378.7505 285.9311	Correlation 0.174259	Signif. Prob 0.1221	Number 80		
	19 626 645 Estimate 58.347947 0.1315542 Mean 89.39787	0.017935 283.3554 70.10861 80 Sum of Squares M 196129.5 6262642.5 6458772.0 Estimate Std Error 58.347947 32.56149 0.1315542 0.084171 Mean Std Dev 89.39787 378.7505	0.017935 283.3554 70.10861 80 Sum of Squares Mean Square 196129.5 196129 6262642.5 80290 6458772.0 6458772.0 Estimate Std Error t Ratio 58.347947 32.56149 1.79 0.1315542 0.084171 1.56 Mean Std Dev Correlation 89.39787 378.7505 0.174259	0.017935 283.3554 70.10861 80 Sum of Squares Mean Square F Ratio 196129.5 196129 2.4428 6262642.5 80290 Prob > F 6458772.0 0.1221 Estimate Std Error t Ratio 58.347947 32.56149 1.79 0.0770 0.1315542 0.084171 1.56 0.1221 Mean Std Dev Correlation Signif. Prob 89.39787 378.7505 0.174259 0.1221		

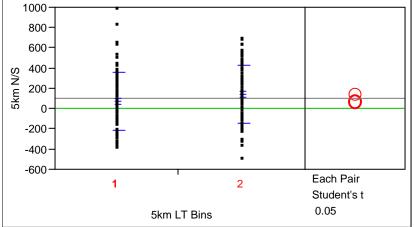


Bivariate Fit of 5km N/S LTB 2 By 5km E/W LTB 2 (Nominal CARP)

Linear Fit

5km N/S LTB 2 = 139.95308 - 0.0283622 5km E/W LTB 2

RSquare RSquare Adj Root Mean Square Error Mean of Response Observations (or Sum Wgts)		0.001289 -0.01119 285.4557 140.3818 82			
Analysis of Variance					
Source DF	Sum of S	quares	Mean Square	F Ratio	
Model 1		8415.6	8415.6	0.1033	
Error 80	651	8795.5	81484.9	Prob > F	
C. Total 81	652	7211.2		0.7488	
Parameter Estimates					
Term	Estimate	Std Error	t Ratio	Prob> t	
Intercept	139.95308	31.55152		<.0001	
5km E/W LTB 2	-0.028362	0.088254	-0.32	0.7488	
Correlation					
Variable	Mean	Std Dev	Correlation	Signif. Prob	Number
5km E/W LTB 2	-15.1166	359.3861	-0.03591	0.7488	82
5km N/S LTB 2	140.3818	283.8712			02



Oneway Analysis of 5km N/S By 5km LT Bins (Nominal CARP)

Means and Std Deviations

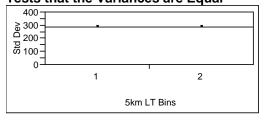
Level	Number	Mean	Std Dev	Std Err Mean	Lower 95%	Upper 95%
1	80	70.109	285.931	31.968	6.478	133.74
2	82	140.382	283.871	31.348	78.008	202.76

Means Comparisons Comparisons for each pair using Student's t

t	Alpha	
1.97490	0.05	
Abs(Dif)-LSD	2	1
2	-87.868	-18.142
1	-18.142	-88.960

Positive values show pairs of means that are significantly different.

Tests that the Variances are Equal

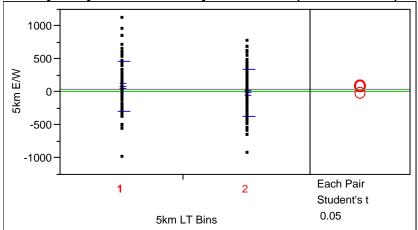


Level 1 2	Count 80 82	Std Dev 285.9311 283.8712	Mean	AbsDif to Mear 229.3837 236.1517	7	MeanAbsDif to Median 229.3326 236.1347
Test O'Brien[.5] Brown-Forsy Levene Bartlett F Test 2-side		F Ratio 0.0048 0.0712 0.0706 0.0042 1.0146	DFNum 1 1 1 1 79	DFDen 160 160 160 81	p-Value 0.9448 0.7899 0.7908 0.9486 0.9479	

Welch Anova testing Means Equal, allowing Std Devs Not Equal

F Ratio	DFNum	DFDen	Prob > F
2.4634	1	159.84	0.1185

t Test 1.5695



Oneway Analysis of 5km E/W By 5km LT Bins (Nominal CARP)

Means and Std Deviations

Level	Number	Mean	Std Dev	Std Err Mean	Lower 95%	Upper 95%
1	80	89.398	378.751	42.346	5.11	173.68
2	82	-15.117	359.386	39.688	-94.08	63.85

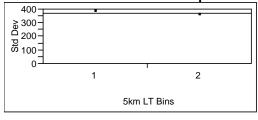
Means Comparisons

Comparisons for each pair using Student's t

t 1.97490	Alpha 0.05	•
Abs(Dif)-LSD	1	2
1	-115.25	-10.03
2	-10.03	-113.83

Positive values show pairs of means that are significantly different.

Tests that the Variances are Equal



Level 1 2	Count 80 82	Std Dev 378.7505 359.3861	Mean	AbsDif to Mear 305.3941 294.4267	l	MeanAbsDif to Median 305.3941 292.4927
Test O'Brien[.5] Brown-Forsy Levene Bartlett F Test 2-side		F Ratio 0.2339 0.1441 0.1079 0.2189 1.1107	DFNum 1 1 1 1 79	DFDen 160 160 160 81	p-Value 0.6293 0.7047 0.7430 0.6399 0.6392	

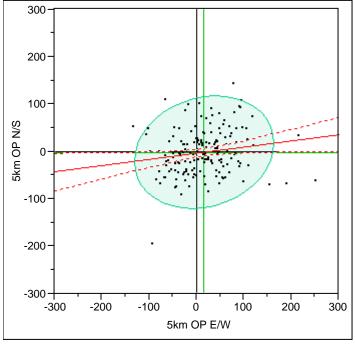
Welch Anova testing Means Equal, allowing Std Devs Not Equal

F Ratio	DFNum	DFDen	Prob > F
3.2430	1	159.05	0.0736

t Test 1.8008

74

Bivariate Fit of 5km OP N/S By 5km OP E/W (Nominal OP)



Linear Fit Bivariate Normal Ellipse P=0.950

Linear Fit

5km OP N/S = -3.859172 + 0.1328734 5km OP E/W

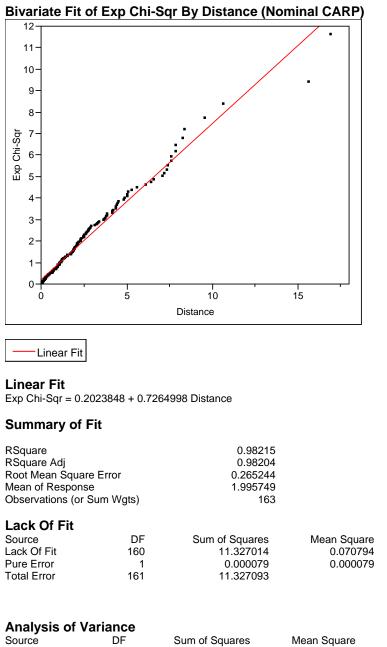
Summary of Fit

RSquare	0.027325
RSquare Adj	0.021246
Root Mean Square Error	47.8902
Mean of Response	-1.77951
Observations (or Sum Wgts)	162

Analysis of Variance

Source Model Error C. Total	DF S 1 160 161	um of Squares 10308.70 366955.36 377264.06	Mean Square 10308.7 2293.5	10308.7 4.4948	
Parameter Estin	nates				
Term	Estim	ate Std Error	t Ratio	Prob> t	
Intercept	-3.859	3.888374	-0.99	0.3225	
5km OP E/W	0.13287	0.062673	2.12	0.0355	
Correlation					
Variable	Mean	Std Dev	Correlation	Signif. Prob	Number
5km OP E/W	15.65148	60.22144	0.165302	0.0355	162
5km OP N/S	-1.77951	48.40718			

Appendix C: 15km JMP 6 Analysis Output



Analysis of	Variance							
Source	DF	Sum of Squares	Mean	Square	F Ratio			
Model	1	623.26073	6	23.261	8858.846			
Error	161	11.32709		0.070	Prob > F			
C. Total	162	634.58782			<.0001			
Parameter Estimates								
Term	Estima	ate Std Error	t Ratio	Prob> t				
Intercept	0.20238	48 0.02819	7.18	<.0001				
Distance	0.72649	98 0.007719	94.12	<.0001				

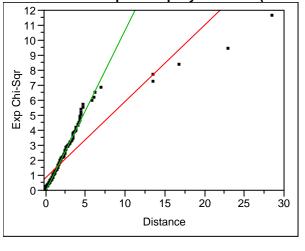
F Ratio

891.8347

Prob > F

0.0267 Max RSq 1.0000





Linear Fit

Linear Fit

Exp Chi-Sqr = 0.8487173 + 0.5074542 Distance

Summary of Fit

RSquare RSquare Adj Root Mean Square Mean of Response Observations (or S		0.793745 0.792464 0.901644 1.995749 163			
Lack Of Fit Source Lack Of Fit Pure Error Total Error	DF 160 1 161	Sum of Squares 130.88671 0.00014 130.88685	(n Square).818042).000143	F Ratio 5728.384 Prob > F 0.0105
Analysis of Va	riance				Max RSq 1.0000
Source	DF	Sum of Squares	Mean Sq		F Ratio
Model	1	503.70097		8.701	619.5875
Error	161	130.88685	C).813	Prob > F
C. Total	162	634.58782			<.0001
Parameter Est	imates				
Term	Estima	ate Std Error	t Ratio	Prob> t	
Intercept	0.84871	73 0.084327	10.06	<.0001	
Distance	0.50745	42 0.020387	24.89	<.0001	

Linear Fit

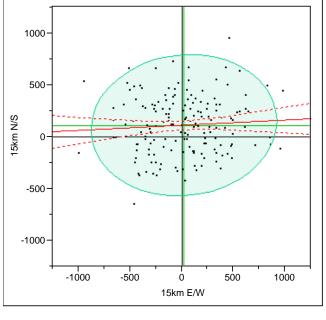
Exp Chi-Sqr = -0.067938 + 1.0699483 Distance

Summary of Fit

RSquare	0.993236
RSquare Adj	0.993193
Root Mean Square Error	0.12467
Mean of Response	1.747532
Observations (or Sum Wgts)	157

Lack Of Fit							
Source	DF	Sum of Squares	Mea	n Square	F Ratio		
Lack Of Fit	154	2.4089644	(0.015643	109.5384		
Pure Error	1	0.0001428	(0.000143	Prob > F		
Total Error	155	2.4091072			0.0760		
					Max RSq		
					1.0000		
Analysis of Va	riance						
Source	DF	Sum of Squares	Mean Sc	luare	F Ratio		
Model	1	353.77827	353	3.778	22761.81		
Error	155	2.40911	C).016	Prob > F		
C. Total	156	356.18737			<.0001		
Parameter Esti	mates						
Term	Estimat	e Std Error	t Ratio	Prob> t			
Intercept	-0.06793	0.015614	-4.35	<.0001			
Distance	1.069948	3 0.007092	150.87	<.0001			



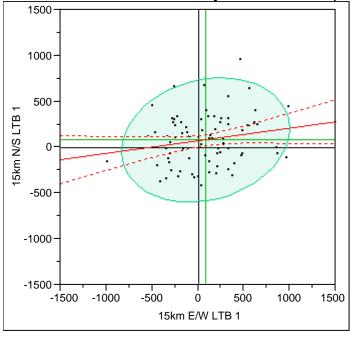


Linear Fit

15km N/S = 111.32849 + 0.0493597 15km E/W

RSquare RSquare Adj Root Mean Square E Mean of Response Observations (or Sur		0.0042 -0.0019 279.744 112.438 10	97 41			
Lack Of Fit Source Lack Of Fit Pure Error Total Error	DF 160 1 161	Sum of Squares 1259934((1259934())	l Square 78745.9 0.0	F Ra Prob > Max R 1.00	> F Sq
Analysis of Var Source Model Error C. Total		Sum of Squares 53276 12599340 12652615		uare 75.6 56.8	F Ratio 0.6808 Prob > F 0.4105	
Parameter Estin Term Intercept 15km E/W	nates Estimate 111.3284 0.049359	9 21.95251	t Ratio 5.07 0.83	Prob> t <.0001 0.4105		
Correlation Variable 15km E/W 15km N/S	Mean 22.48796 112.4385	Std Dev 367.396 279.4684	Correlation 0.064889		. Prob .4105	Number 163

Bivariate Fit of 15km N/S LTB 1 By 15km E/W LTB 1 (Nominal CARP)



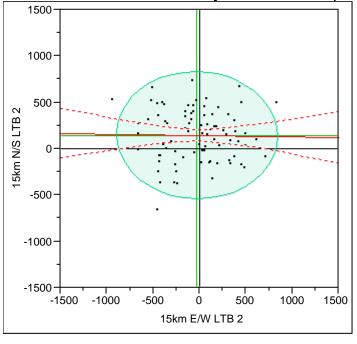
Linear Fit Bivariate Normal Ellipse P=0.950

Linear Fit

15km N/S LTB 1 = 67.491734 + 0.1373675 15km E/W LTB 1

RSquare RSquare Adj Root Mean Square E Mean of Response Observations (or Sur		0.034689 0.022313 273.1787 78.08283 80					
Analysis of Vari	iance						
Source Model Error C. Total	DF 1 78 79	Sum of Squ 209 ⁷ 58208 60300	175.4 374.9	Me	an Square 209175 74627	F Ratio 2.8030 Prob > F 0.0981	
Parameter Estir	nates						
Term		Estimate		Std Error	t Ratio	Prob> t	
Intercept 15km E/W LTB 1		67.491734 0.1373675		31.19057 0.082049	2.16 1.67	0.0335 0.0981	
Correlation Variable 15km E/W LTB 1 15km N/S LTB 1		Mean 77.10047 78.08283	Std 374.5 276.2		Correlation 0.186249	Signif. Prob 0.0981	Number 80

Bivariate Fit of 15km N/S LTB 2 By 15km E/W LTB 2 (Nominal CARP)

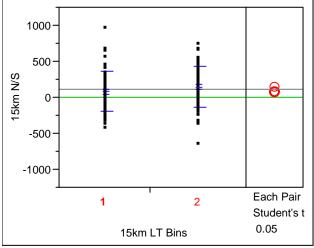


Linear Fit Bivariate Normal Ellipse P=0.950

Linear Fit

15km N/S LTB 2 = 145.11784 - 0.0144124 15km E/W LTB 2

RSquare RSquare Adj Root Mean Square Mean of Response Observations (or Su		0.000333 -0.01201 281.8586 145.5524 83					
Analysis of Va	riance						
Source Model Error C. Total	DF 1 81 82	6434	uares 142.2 986.0 128.2	Me	an Square 2142.2 79444.3	F Ratio 0.0270 Prob > F 0.8700	
Parameter Esti Term Intercept 15km E/W LTB 2	mates	Estimate 145.11784 -0.014412		Std Error 31.05097 0.087769	t Ratio 4.67 -0.16	Prob> t <.0001 0.8700	
Correlation Variable 15km E/W LTB 2 15km N/S LTB 2		Mean -30.1506 145.5524	354.	Dev 6377 1813	Correlation -0.01824	Signif. Prob 0.8700	Number 83



Oneway Analysis of 15km N/S By 15km LT Bins (Nominal CARP)

Means and Std Deviations

Level	Number	Mean	Std Dev	Std Err Mean	Lower 95%	Upper 95%
1	80	78.083	276.278	30.889	16.600	139.57
2	83	145.552	280.181	30.754	84.373	206.73

Means Comparisons Comparisons for each pair using Student's t

t 1.97481	Alpha 0.05	
Abs(Dif)-LSD	2	1
2	-85.305	-18.631
1	-18.631	-86.889

Positive values show pairs of means that are significantly different.

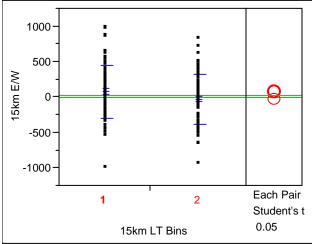
Tests that the Variances are Equal



Level 1 2	Count 80 83	Std Dev 276.2784 280.1813	Mean	AbsDif to Mear 222.6416 232.0265	6	MeanAbsDif to Median 221.5912 231.3428
2	00	200.1010		202.0200	,	201.0420
Test		F Ratio	DFNum	DFDen	p-Value	
O'Brien[.5]		0.0168	1	161	0.8970	
Brown-Forsy	the	0.1502	1	161	0.6989	
Levene		0.1432	1	161	0.7056	
Bartlett		0.0157	1		0.9002	
F Test 2-side	d	1.0285	82	79	0.9013	

Welch Anova testing Means Equal, allowing Std Devs Not Equal

F Ratio	DFNum	DFDen	Prob > F
2.3960	1	160.91	0.1236
t Test 1.5479			



Oneway Analysis of 15km E/W By 15km LT Bins (Nominal CARP)

Means and Std Deviations

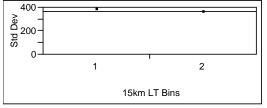
Level	Number	Mean	Std Dev	Std Err Mean	Lower 95%	Upper 95%
1	80	77.100	374.591	41.881	-6.3	160.46
2	83	-30.151	354.638	38.927	-107.6	47.29

Means Comparisons Comparisons for each pair using Student's t

1.97481	0.05	
Abs(Dif)-LSD	1	2
1	-113.83	-5.55
2	-5.55	-111.76

Positive values show pairs of means that are significantly different.

Tests that the Variances are Equal



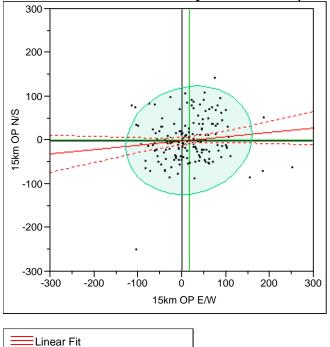
Level 1 2	Count 80 83	Std Dev 374.5915 354.6377	Mean/	AbsDif to Mear 299.8042 289.1837	2	MeanAbsDif to Median 299.8042 289.0077
Test O'Brien[.5] Brown-Forsy Levene Bartlett F Test 2-side		F Ratio 0.2584 0.1041 0.1018 0.2397 1.1157	DFNum 1 1 1 1 79	DFDen 161 161 161 82	p-Value 0.6119 0.7473 0.7501 0.6244 0.6235	

Welch Anova testing Means Equal, allowing Std Devs Not Equal

F Ratio	DFNum	DFDen	Prob > F
3.5185	1	159.66	0.0625
t Test			

1.8758

Bivariate Fit of 15km OP N/S By 15km OP E/W (Nominal OP)



Linear Fit

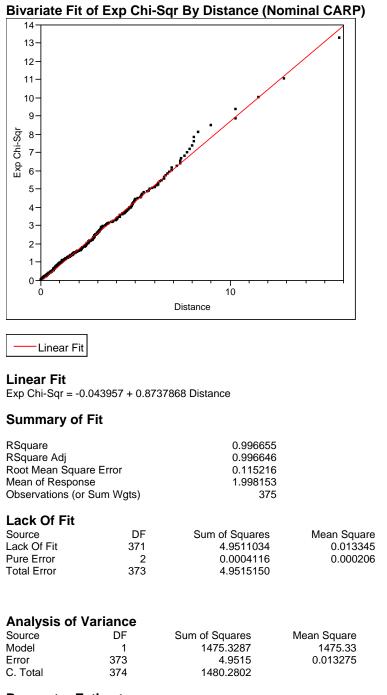
-

15km OP N/S = -1.011746 + 0.0975291 15km OP E/W

Bivariate Normal Ellipse P=0.95

RSquare Rsquare Adj Root Mean Square E Mean of Response Observations (or Su			0.01278 0.006648 50.41698 0.552814 163			
Lack Of Fit Source Lack Of Fit Pure Error Total Error	DF 160 1 161	2	of Squares 409241.40 0.00 409241.40	Mean Squar 2557.7 0.0	6.	
Analysis of Var Source Model Error C. Total	iance DF 1 161 162	4092	quares 297.85 241.40 539.25	Mean Square 5297.85 2541.87	F Ratio 2.0842 Prob > F 0.1508	
Parameter Estin Term Intercept 15km OP E/W	-	Estimate 1.011746 0975291	Std Error 4.094969 0.067556	t Ratio -0.25 1.44	Prob> t 0.8052 0.1508	
Correlation Variable 15km OP E/W 15km OP N/S	M 16.04 0.552	198 58	Std Dev 3.63512 0.58542	Correlation 0.113049	Signif. Prob 0.1508	Number 163

Appendix D: 45km JMP 6 Analysis Output



Parameter Estimates Term Estimate Std Error t Ratio Prob>|t| Intercept -0.043957 <.0001 0.008539 -5.15 Distance 0.8737868 0.002621 333.37 0.0000 F Ratio

64.8420

Prob > F

F Ratio

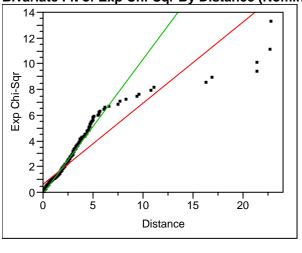
111137.2

Prob > F

0.0000

0.0153 Max RSq 1.0000





Linear Fit	
Linear Fit	

Linear Fit

Exp Chi-Sqr = 0.6325086 + 0.6302233 Distance

Summary of Fit

RSquare RSquare Adj Root Mean Squar Mean of Respons Observations (or S	е	0.8511 0.8507 0.7686 1.9981 3	24 56	
Lack Of Fit Source Lack Of Fit Pure Error Total Error	DF 369 4 373	Sum of Square 220.3799 0.0001 220.3800	5 0.59 4 0.00	uare F Ra 7236 16924. 0035 Prob > <.00 Max R 1.00
Analysis of V	ariance			
Source Model Error C. Total	DF 1 373 374	Sum of Squares 1259.9001 220.3801 1480.2802	Mean Squar 1259.90 0.59	0 2132.419
Parameter Es Term Intercept	s timates Estima 0.63250		t Ratio F 12.78	Prob> t <.0001

0.013648

Linear Fit

Distance

Exp Chi-Sqr = -0.056533 + 1.0449162 Distance

Summary of Fit

RSquare	0.988978
RSquare Adj	0.988947
Root Mean Square Error	0.145563
Mean of Response	1.65801
Observations (or Sum Wgts)	354

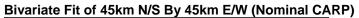
0.6302233

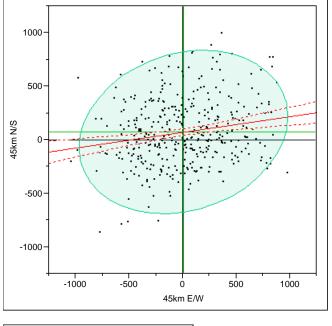
46.18

<.0001

F Ratio 16924.85 Prob > F <.0001 Max RSq 1.0000

Lack Of Fit					
Source	DF	Sum of Squares	Mea	an Square	F Ratio
Lack Of Fit	348	7.4582215		0.021432	607.3446
Pure Error	4	0.0001412		0.000035	Prob > F
Total Error	352	7.4583626			<.0001
					Max RSq
					1.0000
Analysis of Va	ariance				
Source	DF	Sum of Squares	Mean S	quare	F Ratio
Model	1	669.21491	66	69.215	31583.83
Error	352	7.45836		0.021	Prob > F
C. Total	353	676.67327			0.0000
	· · · · · · · ·				
Parameter Es	timates				
Term	Estima	ate Std Error	t Ratio	Prob> t	
Intercept	-0.0565		-4.57	<.0001	
Distance	1.04491	62 0.00588	177.72	0.0000	





Linear Fit	
——Bivariate Normal Ellipse P=0.950	þ

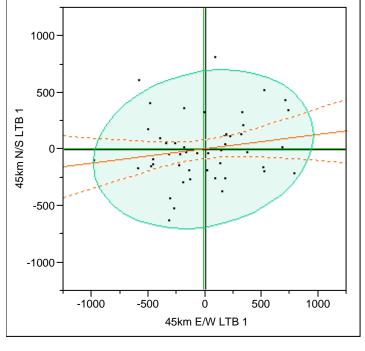
Linear Fit

45km N/S = 72.951882 + 0.1476738 45km E/W

Summary of Fit

RSquare RSquare Adj Root Mean Square Mean of Response Observations (or Su		0.03555 0.0330 306.70 74.1276 37	14 14 51			
Lack Of Fit Source Lack Of Fit Pure Error Total Error	DF 371 2 373	Sum of Squares 35086535 0 35086535	5 9	Square 94572.9 0.0	F Ra Prob : Max R 1.00	> F Sq
Analysis of Van Source Model Error C. Total		um of Squares 1295163 35086535 36381698	Mean Squ 1295 94		F Ratio 13.7687 Prob > F 0.0002	
Parameter Esti Term Intercept 45km E/W	mates Estimate 72.951882 0.1476738	Std Error 15.84116 0.039798	t Ratio 4.61 3.71	Prob> t <.0001 0.0002	İ	
Correlation Variable 45km E/W 45km N/S	Mean 7.961625 74.12761	Std Dev 398.495 311.893	Correlation 0.188678	0	. Prob 0.0002	Number 375





Linear Fit

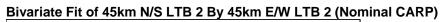
45km N/S LTB 1 = 2.0917632 + 0.1272016 45km E/W LTB 1

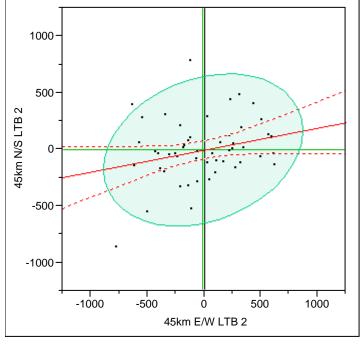
Summary of Fit

Analysis of Variance	
Observations (or Sum Wgts)	48
Mean of Response	0.103331
Root Mean Square Error	284.6842
RSquare Adj	0.010139
RSquare	0.0312

Sum of Squares 120061.9 Source DF Mean Square F Ratio Model 120062 1.4814 1 3728074.7 81045 Prob > F Error 46 C. Total 47 3848136.7 0.2298 **Parameter Estimates** Prob>|t| Term Estimate Std Error t Ratio

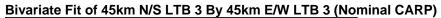
Intercept 45km E/W LTB 1	2.0917632 0.1272016		0.05 1.22	0.9597 0.2298	
Correlation Variable 45km E/W LTB 1 45km N/S LTB 1	Mean -15.6321 0.103331	Std Dev 397.339 286.1385	Correlation 0.176635	Signif. Prob 0.2298	Number 48

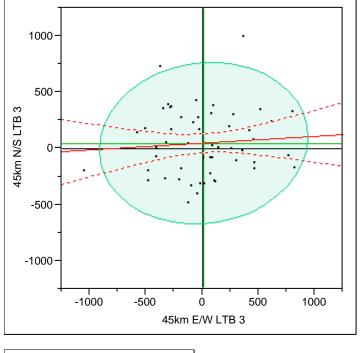




Linear Fit 45km N/S LTB 2 = -4.657001 + 0.1945243 45km E/W LTB 2

RSquare RSquare Adj Root Mean Square Error Mean of Response Observations (or Sum Wgts)		0.064771 0.045685 268.1706 -6.75877 51			
Analysis of Variance					
Source DF	Sum of Sq	uares Me	ean Square	F Ratio	
Model 1		051.7	244052	3.3936	
Error 49	3523	859.1	71915	Prob > F	
C. Total 50	3767	910.9		0.0715	
Parameter Estimates					
Term	Estimate	Std Error	t Ratio	Prob> t	
Intercept	-4.657001	37.56873	-0.12	0.9019	
45km E/W LTB 2	0.1945243	0.105595	1.84	0.0715	
Correlation					
Variable	Mean	Std Dev	Correlation	Signif. Prob	Number
45km E/W LTB 2	-10.8047	359.1552	0.254502	0.0715	51
45km N/S LTB 2	-6.75877	274.5145			





Linear Fit

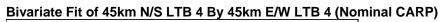
45km N/S LTB 3 = 43.369608 + 0.0618356 45km E/W LTB 3

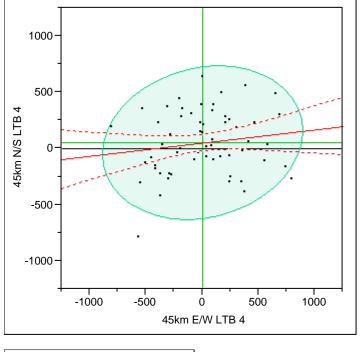
Summary of Fit

RSquare RSquare Adj Root Mean Square Error Mean of Response Observations (or Sum Wgts)		0.006309 -0.01317 296.5201 44.09412 53					
Analysis of	Variance						
Source	DF	Sum of Squares	Mean Square				
Model	1	28472.0	28472.0				
Error	51	4484132.2 879					
C. Total	52	4512604.3					

C. Total	52	45126	604.3		0.5718	
Parameter Esti Term Intercept 45km E/W LTB 3	imates	Estimate 43.369608 0.0618356	Std Error 40.75006 0.108663	t Ratio 1.06 0.57	Prob> t 0.2922 0.5718	
Correlation Variable 45km E/W LTB 3 45km N/S LTB 3		Mean 11.71682 44.09412	Std Dev 378.4157 294.5859	Correlation 0.079432	Signif. Prob 0.5718	Number 53

F Ratio 0.3238 Prob > F

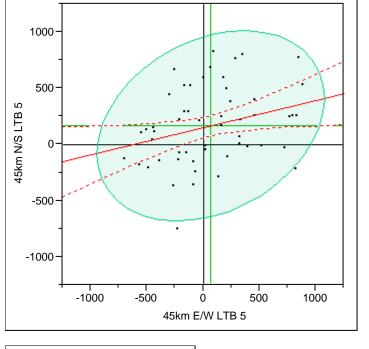




Linear Fit

45km N/S LTB 4 = 48.626014 + 0.1174245 45km E/W LTB 4

RSquare RSquare Adj Root Mean Square Mean of Response Observations (or St			0.023548 0.006713 277.4933 49.32453 60				
Analysis of Va	riance						
Source Model Error C. Total	DF 1 58 59		706.7 146.2	Me	an Square 107707 77003	F Ratio 1.3987 Prob > F 0.2418	
Parameter Est	imates						
Term Intercept 45km E/W LTB 4		Estimate 48.626014 0.1174245		Error 3291 9286	t Ratio 1.36 1.18	Prob> t 0.1800 0.2418	
Correlation Variable 45km E/W LTB 4 45km N/S LTB 4		Mean 5.948603 49.32453	Std Dev 363.862 278.4294		Correlation 0.153455	Signif. Prob 0.2418	Number 60



Bivariate Fit of 45km N/S LTB 5 By 45km E/W LTB 5 (Nominal CARP)

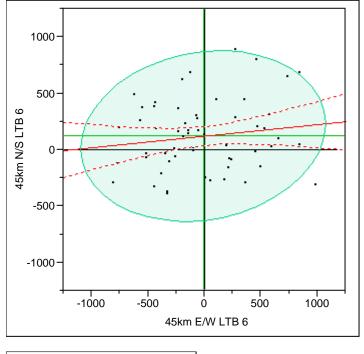
Linear Fit Bivariate Normal Ellipse P=0.950

Linear Fit

45km N/S LTB 5 = 146.66501 + 0.242952 45km E/W LTB 5

RSquare RSquare Adj Root Mean Square Mean of Response Observations (or St			0.084 0.066 333.3 162	6801			
Analysis of Va	riance						
Source	DF	Sum of Sq	uares	Ме	an Square	F Ratio	
Model	1		891.8		524892	4.7223	
Error	51	5668	765.7		111152	Prob > F	
C. Total	52	6193	6193657.4			0.0344	
Parameter Est	imates						
Term		Estimate		Std Error	t Ratio	Prob> t	
Intercept		146.66501		46.38147	3.16	0.0026	
45km Ė/W LTB 5		0.242952		0.111801	2.17	0.0344	
Correlation							
Variable 45km E/W LTB 5 45km N/S LTB 5		Mean 65.74553 162.638	Std 413.5 345.1		Correlation 0.291113	Signif. Prob 0.0344	Number 53
			0.10.1				

Bivariate Fit of 45km N/S LTB 6 By 45km E/W LTB 6 (Nominal CARP)



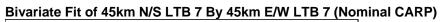
Linear Fit Bivariate Normal Ellipse P=0.950

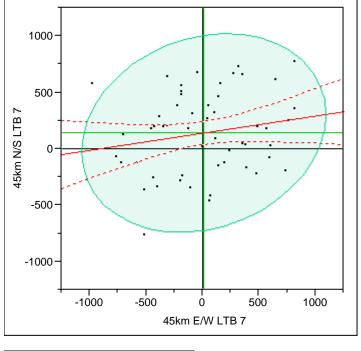
Linear Fit

45km N/S LTB 6 = 121.5945 + 0.1008843 45km E/W LTB 6

Summary of Fit

RSquare RSquare Adj Root Mean Square Er Mean of Response Observations (or Sum			0.020912 0.00311 308.2769 120.3183 57				
Analysis of Varia	ance						
Source	DF	Sum of Sq	uares	Me	an Square	F Ratio	
Model	1	111	637.0		111637	1.1747	
Error	55	5226	906.7		95035	Prob > F	
C. Total	56	5338	543.7			0.2832	
Parameter Estim	ates						
Term		Estimate	Std E	rror	t Ratio	Prob> t	
Intercept		121.5945	40.84	925	2.98	0.0043	
45km E/W LTB 6		0.1008843	0.093	081	1.08	0.2832	
Correlation Variable 45km E/W LTB 6 45km N/S LTB 6		Mean -12.6501 120.3183	Std Dev 442.5745 308.7574		Correlation 0.144608	Signif. Prob 0.2832	Number 57

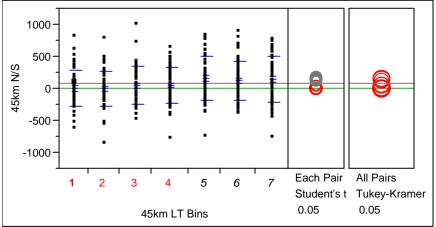




Linear Fit

45km N/S LTB 7 = 137.36229 + 0.1520644 45km E/W LTB 7

RSquare RSquare Adj Root Mean Square Mean of Response Observations (or S	;		0.035013 0.016092 356.2166 138.9278 53				
Analysis of Va	ariance						
Source Model Error C. Total	DF 1 51 52	6471	uares 806.0 403.7 209.7	Mea	an Square 234806 126890	F Ratio 1.8505 Prob > F 0.1797	
Parameter Est	timates						
Term		Estimate			t Ratio	Prob> t	
Intercept 45km E/W LTB 7		137.36229 0.1520644	48.94 0.111		2.81 1.36	0.0071 0.1797	
Correlation Variable 45km E/W LTB 7 45km N/S LTB 7		Mean 10.2948 138.9278	Std Dev 441.9014 359.1178		Correlation 0.187118	Signif. Prob 0.1797	Number 53



Oneway Analysis of 45km N/S By 45km LT Bins (Nominal CARP)

Means and Std Deviations

Level	Number	Mean	Std Dev	Std Err Mean	Lower 95%	Upper 95%
1	48	0.10	286.139	41.301	-82.98	83.19
2	51	-6.76	274.515	38.440	-83.97	70.45
3	53	44.09	294.586	40.464	-37.10	125.29
4	60	49.32	278.429	35.945	-22.60	121.25
5	53	162.64	345.121	47.406	67.51	257.77
6	57	120.32	308.757	40.896	38.39	202.24
7	53	138.93	359.118	49.329	39.94	237.91

Means Comparisons Comparisons for each pair using Student's t t Alpha 1.96643 0.05

Abs(Dif)-LSD	5	7	6	4	3	1	2
5	-117.71	-94.00	-73.30	-0.91	0.84	41.80	50.54
7	-94.00	-117.71	-97.01	-24.62	-22.87	18.09	26.83
6	-73.30	-97.01	-113.50	-41.08	-39.40	1.51	10.29
4	-0.91	-24.62	-41.08	-110.63	-108.99	-68.12	-59.32
3	0.84	-22.87	-39.40	-108.99	-117.71	-76.74	-68.00
1	41.80	18.09	1.51	-68.12	-76.74	-123.68	-114.99
2	50.54	26.83	10.29	-59.32	-68.00	-114.99	-119.99

Positive values show pairs of means that are significantly different.

Level				Mean
5	Α			162.6380
7	Α	В		138.9278
6	Α	В		120.3183
4	Α	В	С	49.3245
3		В	С	44.0941
1			С	0.1033
2			С	-6.7588

Level	- Level	Difference	Lower CL	Upper CL	p-Value Difference
5	2	169.3968	50.542	288.2513	0.0053
5	1	162.5347	41.802	283.2672	0.0085
7	2	145.6865	26.832	264.5410	0.0164
7	1	138.8244	18.092	259.5569	0.0243
6	2	127.0771	10.286	243.8686	0.0330
6	1	120.2150	1.513	238.9171	0.0472
5	3	118.5439	0.838	236.2500	0.0484

Level 5	- Level 4	Difference 113.3135	Lower CL -0.908	Upper CL 227.5349	p-Value Difference
7	3	94.8336	-22.872	212.5398	0.1140
7	4	89.6032	-24.618	203.8247	0.1238
6	3	76.2242	-39.398	191.8468	0.1957
6	4	70.9938	-41.079	183.0670	0.2137
4	2	56.0833	-59.321	171.4878	0.3399
3	2	50.8529	-68.002	169.7074	0.4007
4	1	49.2212	-68.117	166.5589	0.4100
3	1	43.9908	-76.742	164.7233	0.4741
5	6	42.3197	-73.303	157.9424	0.4721
5	7	23.7103	-93.996	141.4164	0.6923
7	6	18.6095	-97.013	134.2321	0.7518
1	2	6.8621	-114.990	128.7144	0.9119
4	3	5.2304	-108.991	119.4518	0.9283

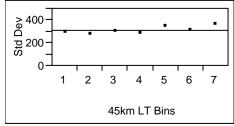
	0.05					
5	7	6	4	3	1	2
-177.46	-153.75	-132.00	-58.90	-58.92	-19.49	-9.80
-153.75	-177.46	-155.71	-82.61	-82.63	-43.20	-33.51
-132.00	-155.71	-171.12	-97.98	-98.10	-58.75	-49.01
-58.90	-82.61	-97.98	-166.79	-166.98	-127.69	-117.91
-58.92	-82.63	-98.10	-166.98	-177.46	-138.03	-128.34
-19.49	-43.20	-58.75	-127.69	-138.03	-186.48	-176.85
-9.80	-33.51	-49.01	-117.91	-128.34	-176.85	-180.91
	-177.46 -153.75 -132.00 -58.90 -58.92 -19.49	0.05 5 7 -177.46 -153.75 -153.75 -177.46 -132.00 -155.71 -58.90 -82.61 -58.92 -82.63 -19.49 -43.20	0.05 5 7 6 -177.46 -153.75 -132.00 -153.75 -177.46 -155.71 -132.00 -155.71 -171.12 -58.90 -82.61 -97.98 -58.92 -82.63 -98.10 -19.49 -43.20 -58.75	0.05 5 7 6 4 -177.46 -153.75 -132.00 -58.90 -153.75 -177.46 -155.71 -82.61 -132.00 -155.71 -171.12 -97.98 -58.90 -82.61 -97.98 -166.79 -58.92 -82.63 -98.10 -166.98 -19.49 -43.20 -58.75 -127.69	$\begin{array}{cccccccccccccccccccccccccccccccccccc$	0.05 5 7 6 4 3 1 -177.46 -153.75 -132.00 -58.90 -58.92 -19.49 -153.75 -177.46 -155.71 -82.61 -82.63 -43.20 -132.00 -155.71 -171.12 -97.98 -98.10 -58.75 -58.90 -82.61 -97.98 -166.79 -166.98 -127.69 -58.92 -82.63 -98.10 -166.98 -177.46 -138.03 -19.49 -43.20 -58.75 -127.69 -138.03 -186.48

Positive values show pairs of means that are significantly different.

Level		Mean
5	А	162.6380
7	А	138.9278
6	A	120.3183
4	Α	49.3245
3	A	44.0941
1	А	0.1033
2	А	-6.7588

Level	- Level	Difference	Lower CL	Upper CL Difference
5	2	169.3968	-9.797	348.5905
5	1	162.5347	-19.490	344.5598
7	2	145.6865	-33.507	324.8802
7	4	138.8244	-43.201	320.8495
1	1			
6	2	127.0771	-49.006	303.1605
6	1	120.2150	-58.749	299.1790
5	3	118.5439	-58.918	296.0062
5	4	113.3135	-58.895	285.5221
7	3	94.8336	-82.629	272.2960
7	4	89.6032	-82.605	261.8118
6	3	76.2242	-98.097	250.5453
6	4	70.9938	-97.976	239.9635
4	2	56.0833	-117.909	230.0755
3	2	50.8529	-128.341	230.0466
4	1	49.2212	-127.686	226.1281
3	1	43.9908	-138.034	226.0159
5	6	42.3197	-132.001	216.6409
5	7	23.7103	-153.752	201.1726
7	6	18.6095	-155.712	192.9306
1	2	6.8621	-176.851	190.5756
4	3	5.2304	-166.978	177.4390

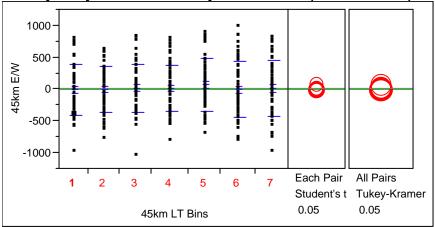
Tests that the Variances are Equal



Level	Count	Std Dev	MeanA	bsDif to Mear	า	MeanAbsDif to Median
1	48	286.1385		216.4796	5	213.5568
2	51	274.5145		192.9826	5	192.0040
3	53	294.5859		238.2328	3	236.6768
4	60	278.4294		228.222	5	226.7965
5	53	345.1214		279.0342	2	278.9815
6	57	308.7574		247.664	5	247.8394
7	53	359.1178		297.026	1	294.7004
Test		F Ratio	DFNum	DFDen	Prob > F	:
O'Brien[.5]		1.2329	6	368	0.2885	5
Brown-Forsythe	е	1.9188	6	368	0.0768	3
Levene		1.9868	6	368	0.0667	•
Bartlett		1.1944	6		0.3057	,

Welch Anova testing Means Equal, allowing Std Devs Not Equal

F Ratio	DFNum	DFDen	Prob > F
2.4046	6	162.36	0.0297



Oneway Analysis of 45km E/W By 45km LT Bins (Nominal CARP)

Means and Std Deviations

Level	Number	Mean	Std Dev	Std Err Mean	Lower 95%	Upper 95%
1	48	-15.632	397.339	57.351	-131.0	99.74
2	51	-10.805	359.155	50.292	-111.8	90.21
3	53	11.717	378.416	51.979	-92.6	116.02
4	60	5.949	363.862	46.974	-88.0	99.94
5	53	65.746	413.535	56.803	-48.2	179.73
6	57	-12.650	442.575	58.620	-130.1	104.78
7	53	10.295	441.901	60.700	-111.5	132.10

Means Comparisons Comparisons for each pair using Student's t t Alpha 1 96643 0.05

Abs(Dif)-LSD	5	3	7	4	2	6	1
5	-153.14	-99.11	-97.69	-88.81	-78.08	-72.03	-75.70
3	-99.11	-153.14	-151.72	-142.84	-132.11	-126.06	-129.73
7	-97.69	-151.72	-153.14	-144.26	-133.53	-127.48	-131.15
4	-88.81	-142.84	-144.26	-143.93	-133.39	-127.21	-131.08
2	-78.08	-132.11	-133.53	-133.39	-156.11	-150.10	-153.71
6	-72.03	-126.06	-127.48	-127.21	-150.10	-147.67	-151.45
1	-75.70	-129.73	-131.15	-131.08	-153.71	-151.45	-160.92

Positive values show pairs of means that are significantly different.

Level		Mean
5	А	65.74553
3	А	11.71682
7	A	10.29480
4	А	5.94860
2	A	-10.80468
6	А	-12.65008
1	А	-15.63213

Level	- Level	Difference	Lower CL	Upper CL	p-Value Difference
5	1	81.37766	-75.698	238.4538	0.3090
5	6	78.39561	-72.032	228.8237	0.3061
5	2	76.55021	-78.083	231.1830	0.3310
5	4	59.79693	-88.808	208.4020	0.4293
5	7	55.45073	-97.688	208.5895	0.4769
5	3	54.02871	-99.110	207.1675	0.4883
3	1	27.34895	-129.727	184.4251	0.7323

Level	- Level	Difference	Lower CL	Upper CL	p-Value Difference
7	1	25.92693	-131.149	183.0030	0.7457
3	6	24.36690	-126.061	174.7950	0.7503
7	6	22.94488	-127.483	173.3730	0.7644
3	2	22.52150	-132.111	177.1543	0.7747
4	1	21.58074	-131.079	174.2402	0.7812
7	2	21.09948	-133.533	175.7323	0.7886
4	6	18.59869	-127.212	164.4089	0.8021
4	2	16.75328	-133.391	166.8976	0.8264
3	4	5.76822	-142.837	154.3733	0.9392
2	1	4.82746	-153.706	163.3605	0.9523
7	4	4.34620	-144.259	152.9513	0.9542
6	1	2.98205	-151.453	157.4167	0.9697
2	6	1.84541	-150.103	153.7942	0.9810
3	7	1.42202	-151.717	154.5608	0.9854

Comparisons for all pairs using Tukey-Kramer HSD

2.96474		0.05					
Abs(Dif)-LSD	5	3	7	4	2	6	1
5	-230.88	-176.85	-175.43	-164.25	-156.59	-148.40	-155.44
3	-176.85	-230.88	-229.46	-218.28	-210.61	-202.43	-209.47
7	-175.43	-229.46	-230.88	-219.70	-212.04	-203.85	-210.89
4	-164.25	-218.28	-219.70	-217.00	-209.62	-201.24	-208.58
2	-156.59	-210.61	-212.04	-209.62	-235.37	-227.24	-234.19
6	-148.40	-202.43	-203.85	-201.24	-227.24	-222.63	-229.85
1	-155.44	-209.47	-210.89	-208.58	-234.19	-229.85	-242.61

Positive values show pairs of means that are significantly different.

Level		Mean
5	А	65.74553
3	А	11.71682
7	А	10.29480
4	А	5.94860
2	А	-10.80468
6	А	-12.65008
1	А	-15.63213

Level	- Level	Difference	Lower CL	Upper CL Difference
5	1	81.37766	-155.442	318.1971
5	6	78.39561	-148.401	305.1920
5	2	76.55021	-156.585	309.6859
5	4	59.79693	-164.251	283.8448
5	7	55.45073	-175.432	286.3339
5	3	54.02871	-176.854	284.9118
3	1	27.34895	-209.470	264.1683
7	1	25.92693	-210.892	262.7463
3	6	24.36690	-202.429	251.1633
7	6	22.94488	-203.851	249.7413
3	2	22.52150	-210.614	255.6572
4	1	21.58074	-208.580	251.7412
7	2	21.09948	-212.036	254.2352
4	6	18.59869	-201.235	238.4328
4	2	16.75328	-209.615	243.1217
3	4	5.76822	-218.280	229.8161
2	1	4.82746	-234.189	243.8435
7	4	4.34620	-219.702	228.3941
6	1	2.98205	-229.855	235.8189
2	6	1.84541	-227.244	230.9345
3	7	1.42202	-229.461	232.3051

Tests that the Variances are Equal

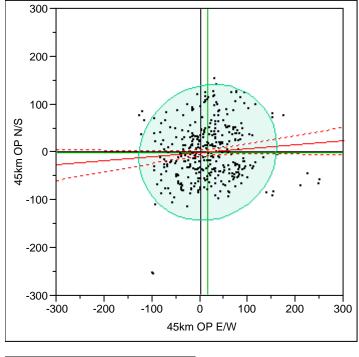
× 400 Std Dev 200 200 V 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1			-	•			•
Ū	1	2	3	4	5	6	7
			45ki	m LT	Bins		

Level	Count	Std Dev	MeanA	AbsDif to Mear	า	MeanAbsDif to Median
1	48	397.3390		326.6421	1	326.6421
2	51	359.1552		299.1003	3	298.4077
3	53	378.4157		295.0329	9	294.6817
4	60	363.8620		294.2648	3	294.1885
5	53	413.5354		338.0630)	335.4860
6	57	442.5745		366.6649	9	360.2060
7	53	441.9014		362.9761	1	362.5760
Test		F Ratio	DFNum	DFDen	Prob > F	;
O'Brien[.5]		1.0776	6	368	0.3753	•
Brown-Forsyth	е	0.8953	6	368	0.4983	i
Levene		1.0443	6	368	0.3961	
Bartlett		0.7998	6		0.5699)

Welch Anova testing Means Equal, allowing Std Devs Not Equal

F Ratio	DFNum	DFDen	Prob > F
0.2408	6	162.4	0.9624

Bivariate Fit of 45km OP N/S By 45km OP E/W (Nominal OP)



Linear Fit -Bivariate Normal Ellipse P=0.950

Linear Fit 45km OP N/S = -1.500542 + 0.0839099 45km OP E/W

Summary of Fit

RSquare RSquare Adj Root Mean Square Error Mean of Response Observations (or Sum Wgts)			0.007213 0.004551 57.91268 -0.13898 375		
Lack Of Fit Source	DF	Sum	n of Squares	Mean Square	
Lack Of Fit Pure Error	371 2		1250996.5 0.0	3371.90 0.00	
Total Error	373	1250996.5			Max RSq 1.0000
Analysis of Var	iance				
Source Model Error C. Total	DF 1 373 374	12	Squares 9088.7 50996.5 60085.2	Mean Square 9088.70 3353.88	F Ratio 2.7099 Prob > F 0.1006
Parameter Estin Term Intercept 45km OP E/W	mates	Estimate -1.500542 0.0839099	Std Error 3.102866 0.050973	t Ratio -0.48 1.65	Prob> t 0.6290 0.1006
Correlation Variable		Mean	Std Dev	Correlation	Signif. Prob

Number

Variable	Mean	Std Dev	Correlation	Signif. Prob	Number
45km OP E/W	16.2265	58.74914	0.084928	0.1006	375
45km OP N/S	-0.13898	58.04491			

Appendix E: Bivariate Normal MATLAB Code

```
function [M] = BivarStats(Z)
D = [];
k = length(Z);
x = Z(:, 2);
y = Z(:, 1);
Zbar = [mean(x) mean(y)]';
Sigma = cov([x,y])
v = [];
for i = 1:k
    xt = (i - .5)/k;
    Zi = Z(i,:)';
    di2 = (Zi - Zbar)'*(Sigma^-1)*(Zi - Zbar);
    Xsq = chi2inv(xt,2);
    D = [D;di2,Xsq];
end
M = sort(D);
plot(M(:,1),M(:,2),'*')
mu = [mean(x) mean(y)]
x1 = -1250:10:1250;
x2 = -1250:10:1250;
[X1, X2] = meshgrid(x1, x2);
F = mvnpdf([X1(:) X2(:)], mu, Sigma);
F = reshape(F, length(x2), length(x1));
h = surf(x1, x2, F);
caxis([min(F(:))-.5*range(F(:)),max(F(:))]);
axis([-1250 1250 -1250 1250 0 2.0e-6])
xlabel('x1'); ylabel('x2'); zlabel('Probability Density');
[V,D] = eigs(Sigma)
```

Note: This script is optimized to view CARP data. In order to view OP data, variables x1, x2, and *axis* must be changed to smaller scales to ensure correct plotting of data.

Appendix F: Bivariate Normal MATLAB Input Script

format compact

if r == 1

Ζ	=	[150.437523	
		-131.954659	-92.498073
		-29.42237	946.396638
		159.860508	5.546833
		979.366917	469.531369
		32.588577	325.299697
		-46.562242	403.046916
		32.588205	-303.228644
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13.779812	63.218873
13.500566	-25.176625
-6.796856	-34.825823
-54.747081	54.900743
-31.377077	56.786116
13.500566	-25.176625
-6.796856	-34.825823
-54.747081	54.900743

-21.414419	88.284619
-6.424337	44.031387
-54.468219	-6.543553
93.571512	91.279627
-14.152291	-47.802325
6.051884	44.807761
-22.811277	-39.151305
-27.466798	65.215278
6.23809 40.	48226
49.999105	-46.804006
-9.683136	-65.658884
48.229147	106.252177
-67.782402	190.654977]

disp('5km Resolution OP')

Ζ	=	[95.	899	074	87	. 9	952	23	42	2
		13.5	936	05	25	.5	509	3	76	,
		35.8	461	8	81	. 0)75	64	83	5
		-5.2	139	39	36	. 6	500	13	98	}
		79.0	475	17	80	. 6	532	21	02	2
		-39.								
		-84.								3
		106.								
		-70.								
		-76.								-
		-56.								
		79.8								08
		33.5			8.					_
		-20.								
		-4.6			-6					
		66.3								
		-56.								
		-71.								6
		-11.			2.				2	
		5.67							2.0	
		-52.	419	142						
		-70. -21.	01/	246	-3 -3					
		29.6	913	340	-3 10					
		29.0			10 23					
		12.1			23 61					
		-18.	248	91	7.					,
		-49.								
		-48.			71					
		-42.	550	326	-3					
		-22.	066	442	50					
		-39.	104	866	90					•
		36.4	050	35	69					2
		44.2	257	65	92					
		-21.			33					
		-10.	055	529	27					
		-15.					951			

-88.173013	29.391623
86.309948	12.200516
-28.397669 4.28289 4.65	8233
-5.213974 -16.852446	102.370197
15.362606 6.5175 11.0	
69.457627	-56.453311 77.304719
	252.210153 -4.436391
11.824559 -54.468002	14.085758
-10.707359 31.004675	-46.249542
97.111265 -54.653791	-26.06343
3.258755 -15.828247	27.94939
72.158018	28.06057
3.072501	-54.346047
87.985689	27.506152
9.776251	-32.60762
	41.5917 8.872827
24.114596	-65.658869
37.429045	-16.525561
-36.404955	-54.345973
27.187422	15.638387
-35.846266	111.021254
36.311763	70.317187
-44.691461	51.906115
70.481742	-4.103425
51.207416	185.885828
142.73328	75.42004
-84.262623	154.16563
49.160568	-44.142171
74.113393	68.764656
-6.610645	3.881862
20.949185	-2.661824
-11.172825	99.819261
-9.776303	-92.277353
14.0591 0.00	0001
	48.024223 62.331675
17.690366	-17.856445 16.858379
32.214781	-104.920852 -98.377391
-40.315437	-12.97642
-49.347028	-53.347721
13.314294	78.191759
-64.989488	-81.297015
-29.328609	45.362353

12.103942	-61.554863
-8.19334	42.145915
-12.941824	23.956627
-249.341902	92.388278
19.924811	-103.697919
-23.09062	-0.443621
0 0.887282	57.451552
-18.435389 -33.79782 -16.479931 49.345996 26.069776 0.279317	-12.643699 -5.878227 39.927762 31.942177 60.557153 46.027808 53.569674
-44.691605	-6.432693
16.200659	-13.863772
-6.982951	-26.285738
68.898274	100.595888
24.300845	46.582353
46.832965	47.136982
-5.213939	36.600398
78.116919	22.404192
-36.59131	-3.105419
-61.450926	-22.403683
108.934453	52.793912
2.979443	-38.042223
-46.088252	0.33284
-7.262367	-60.667921
82.68019	-73.75498
5.400157	16.969276
-24.207937	-60.889717
38.17362	28.06038
55.118876	37.487838
-47.391866	-82.406224
-33.984041	-15.08374
-15.641955	2.77277
-3.444918	11.645582
-38.17386	3.549204
-68.899703	-37.154699
35.381057	-103.035586
6.517376	106.030249
71.225877	63.330046
11.452151	55.122419
-5.400194	-20.962042
-11.266055	-9.649187
-51.302085	60.002597
-25.97685	63.662538
-24.673545	11.978341
-46.832977	39.373263

33.61178	78.413623
36.311881	39.040486
-19.55256	-6.654597
-21.41451	46.693257
-4.84153	43.587744
-50.09195	30.278634
90.778714	50.464609
-12.476383	-64.32796
-9.683048	11.091034
-8.3796 -31.	609429
-23.276901	84.624566
10.893446	36.378583
30.446377	-45.140433
-12.755752	-75.418985
61.543476	89.615708
-71.413653	183.778566]

disp('15km Resolution OP')

Ζ =	$\begin{bmatrix} 123.365245 \\ 15.176372 \\ 26.349263 \\ 8.286477 \\ 66.292295 \\ -55.864393 \\ 120.573459 \\ -86.125222 \\ -79.234433 \\ -65.7338 \\ 59.868695 \\ 31.749483 \\ -29.79426 \\ -18.621377 \\ 63.777749 \\ -32.494428 \\ -13.500519 \\ 12.662508 \\ -75.230457 \\ -75.603304 \\ -6.517464 \\ 59.774092 \\ 106.605927 \\ -3.258723 \\ -21.9733 \\ -52.884692 \\ -57.819535 \\ -61.358192 \\ -38.080906 \\ -49.439734 \\ 45.250205 \\ 39.384276 \\ \end{bmatrix}$	57.562453 83.7373 -8.096448 24.400489 31.942322 -4.324753 -62.663922 14.751391 -17.190872 -80.964314 -16.08194 -20.407448 -23.845695 38.042442 -25.176581 -13.087405 -18.18928 0.776663 -17.41262 -8.872821 86.621127 74.643222 58.671542 -9.64917 98.488476 69.319098 -45.029378 50.242431 87.951982 79.079132 75.751803
	39.384276	
	JJ.01000	JJ. TTJJ2/

-19.086922	34.936759
95.155205	0.333197
-40.036181	-70.982502
13.500514	-14.751059
-0.931064	-5.212784
-11.452179	132.205056
-35.008352	94.051977
24.207745	1.220043
8.193426	13.863788
-17.224854	14.972903
-3.444953	-40.482254
56.422823	53.237101
-65.54733	86.0666
-66.385396	247.773763
31.656266	-4.21454
27.280255	33.051305
37.428849	29.50221
50.463626	75.197311
-51.394894	68.209964
-31.935857	82.739118
5.213953	56.786069
89.476321	-13.197911
87.241198	-47.247393
76.626706	-125.882896
-73.089263	24.067806
-36.311759	4.76921
-4.562206	163.038113
-40.129004	41.03689
63.778352	31.942372
15.362632	-81.408147
107.910669	11.42436
-4.469122	-38.70769
109.120713	15.528055
19.55264	15.194727
30.725108	-48.800485
-114.057168	-25.508694
37.428954	7.209241
-25.604543	-67.766152
40.036679	-83.29353
-2.141468	-7.98554
-33.518587	110.688514
-0.55864	46.249588
11.172899	27.616666
-32.587404	0.332786
-25.604364	78.524517
-252.600555	-99.483244
30.166634	-31.831207
-37.429029	88.950118
15.269434	25.731201
-109.216277	-93.164014
16.014435	-15.083786
75.323523	-34.270988
-96.552531	38.486346
-30.352905	17.745693
76.811218	173.907681

-72. -41. -42. 103. 11.6 25.7 125. 15.8 9.77 -83. 113.5 24.0 35.8 86.9 -41. -72. 124. -85. -79. -56. 37.3 48.0 -18. -3.11 76.9 -51. -68. -6.7 34.2 45.6 -57. -80. -19. 45.6 -57. -80. -19. 45.6 -57. -80. -19. 45.6 -57. -80. -19. 45.6 -57. -80. -19. 45.6 -57. -80. -19. 45.6 -57. -80. -19. 45.6 -57. -80. -19. 45.6 -57. -80. -19. 45.6 -57. -80. -19.5 -35. -35. -27. -35. -35. -35. -27. -35. -35. -27. -35. -35. -35. -11. -90.	902979 153302 549799 161597 38339 90376 508431 28139 6162 42465 868539 93607 21615 4609 61669 2465 903065 297523 939156 607006 329878 36623 43112 993858 65614 9883 860862 713117 96768 63238 22464 540044 072627 180117 35818 48934 6232 89983 433887 536591 101696 466669 832815 97807 69418 36729 079732 593788	$\begin{array}{c} -8.872754\\ 74.864712\\ 31.94225\\ 50.131544\\ 10.315206\\ 7.763728\\ 73.533563\\ 77.305277\\ 42.478654\\ -2.994573\\ 154.498354\\ 85.62342\\ 24.400273\\ 73.311727\\ 20.185739\\ 70.650243\\ 42.145995\\ -11.867127\\ 12.311838\\ -79.633194\\ -2.66152\\ -3.770786\\ -122.001224\\ 2.883786\\ -29.280298\\ -16.525633\\ 38.042538\\ -49.133118\\ -40.482012\\ -1.774562\\ -12.643713\\ -37.15484\\ 14.640329\\ -37.154614\\ -32.163964\\ 89.837461\\ 11.978886\\ 49.576903\\ -3.549124\\ 48.800764\\ 72.202741\\ -36.378507\\ 49.909665\\ 72.97908\\ 68.986258\\ 89.282837\\ 22.292977\\ 60.66807\\ 29.169412\\ 32.164402\\ \end{array}$
12.5 53.5 -11. -90. 115. -20. 26.1 -8.3	69418 36729 079732	22.292977 60.66807 29.169412

-37.056766	80.1882
43.94633	9.538385
8.193426	13.863788
-12.755736	
-20.018141	-70.64983
66.385299	48.800752
-61.450781	61.000849
-58.471271	250.657379
-32.587614	-73.977107
22.345592	-0.665436
27.09417	-12.976466
58.657201	21.184043
	38.264174
47.018712	70.982702
	61.000764
-23.463097	75.197194
-79.886411	-34.381858
48.695052	-50.796791
-9.310659	25.842101
87.80041	-15.970684
66.10609	-71.647824
85.657989	-119.782754
-69.550971	53.791737
-37.708358	6.765601
-68.992607	-35.823777
-32.587522	51.684246
-11.917782	39.927706
-83.889762	-2.550574
-29.328786	111.243051
-34.44952	33.273148
58.191925	32.164157
9.869312	-66.546175
113.31078	
4.376013	-34.93674
103.813437	35.27003
4.469152	12.643772
24.673265	-77.304449
-64.896288	-38.485649
43.294829	-24.954718
-21.042264	-60.778815
-46.554262	-79.855283
18.155978	18.632944
-21.600854	116.788546
20.018072 22.438828	35.3804 61.222501
-16.75922	9.5383
-21.321471	69.984414
-253.531586	-97.819565
30.445917	-19.852895
-33.704729	94.051978
13.686648	10.869219
-105.677768	-57.672768
-7.914079	-29.945774
90.685996	-15.748838
-93.94532	59.115633

0 41.36953	38
81.001106	151.725649
-34.63586	-2.439965
-81.934504	60.446447
-78.023932 -36.497935	0.887596 29.613115
83.888383	38.597149
30.911463	-3.105439
17.224665	47.691443
-55.491982	9.982084
125.415342	75.863442
-1.862134 6.331207	-13.531055 -22.95843
-89.942221	152.280206
100.554307	93.941546
13.593618	16.636553
38.173852	81.630044
48.13612	31.942284
98.878861 -45.622641	116.45631 20.74033
-79.327695	-39.040095
153.625864	30.501546
-75.604052	-72.091382
-74.113571	8.651285
-49.905688	-44.363985 -95.493668
38.174497 51.487979	-95.493668 18.854886
-5.493298	-41.036805
26.163068	-35.934899
72.157552	5.102142
-59.495865	-80.187952
-75.417037 11.824543	-66.324059 27.39485
33.052838	-8.540037
50.55727	-62.331451
-55.119346	0.554708
-73.927535	-39.261958
-24.021748	-51.462342
37.521675 97.761337	104.255754 30.611736
-18.621453	18.743856
-27.094225	-20.185634
-36.311941	26.729445
-53.350342	78.524626
-9.031352 -33.518714	38.485872 30.278564
-41.89813	76.306366
19.180202	41.591371
40.687838	60.668009
29.514827	31.942209
52.23318	69.319065
-4.376007 -82.679543	24.400264 42.811718
127.369798	29.170243
-4.376001	-20.296583

-21. -19. -12. 41.7 -1.3 11.9 -29. 34.22 -47. -44.7 -10. 14.7 37.5 78.22 -49.3 34.44 -61. -27.5 -50.39.6	087045 662599 11769 96609 17721 328977 63461 019054 691538 521154 1085 21994 09316 905601 49441 636783 746086	7.098309 -61.555185 91.722814 91.057345 9.982016 -3.88186 27.727578 -94.717337 25.842156 82.406453 224.593389 -83.182709 -2.329105 31.720416 56.564566 38.264174 46.360563 44.14249 60.224318 14.97302 -14.640078
99.2 69.8 65.2 -45. -45. -45. -45. -5.9 -18. -49.2 51.2 51.2 5.40 12.58 -54.3 -54.3 -54.3 -21.6 -54.7 -49.2 -21.6 -49.2 -54.7 -49.2 -54.7 -29.6 -49.2 -54.7 -29.6 -36.7 -29.7 -36.7 -29.7 -36.7 -94.37 -94.5 -94	54.00400 52751 3054 67641 757907 250094 833092 5882 156028 519713 9138 905314 08963 0164 755596 69469 92087 707426 00781 188788 87972 070096 626666 5206 831869 4367 45653 273124 497919 18044 049442 9547 504754 5175	-27.061601 -94.273496 -116.455603 32.386032 -7.098153 -46.914937 65.7698 22.514802 12.42214 81.186321

$\begin{array}{c} -84.6\\ -77.8\\ -78.3\\ -78.3\\ -78.3\\ -278.3\\ -27.3\\ -12.5\\ -91.2\\ -32.5\\ -91.2\\ -32.5\\ -31.2\\ -2.5\\ -31.2\\ -2.5\\ -31.2\\ -2.5\\ -$	90628 615309 955163 480981 33278 21204 52325 242996 832644 3674 3941 355866 29874 34555 32524 29011 7889 473957 450803 335746 449914 709641 484916 66367 56802 483635 76474 05546 434265 199253 62428 8047 1239 988754 845153 81865 04314 92341 893563 383276 32667 726421 667466 070132 970701 7685 08962 69664 57324 10653	-14.861527 61.777397 152.612859 -7.652737 -18.521696 24.622181 41.037206 -30.05668 46.360524 -4.43634 66.214227 1.109105 -34.04946 138.638193 100.374254 17.745646 73.866326 39.151428 30.500636 13.531119 -2.994383 33.828668 -13.309172 19.409515 -49.7986 -69.096747 2.772862 -59.226072 -56.453326 -10.203557 -87.840702 -44.918441 10.536483 0.554616 -19.076413 -9.205336 -19.963471 -67.544313 109.912166 36.157261 7.098264 -24.067524 -22.071087 70.539111 -0.554453 22.514821 67.322646 80.964529 58.228033 43.809588 54.900719 24.067537
50.5 -9.3 -51. 118.	57324 10653 488299 710864	54.900719

24.02157	-1.219983
-10.800447	-44.031378
-30.259937	95.604703
-28.863301	83.51548
43.015169	30.833158
-9.962542	-11.867394
10.707323	31.609437
-39.477737	-95.382763
42.922444	23.623985
-43.480994	80.742783
-69.365094	210.175198]

disp('45km Resolution OP')

end

[M] = BivarStats_f(Z);

Appendix G: JPADS-MP Help

JPADS-MP Operation – N Mission

The JPADS-MP is developed by Draper Labs and Planning Systems Inc (PSI), and a complete user's manual is available from them. The following sequence of figures will provide the reader with sufficient familiarity with the JPADS-MP Graphical User Interface (GUI) to recreate the steps taken in this research. Upon starting the JPADS-MP, the user is presented with the main GUI page as shown in following figure:

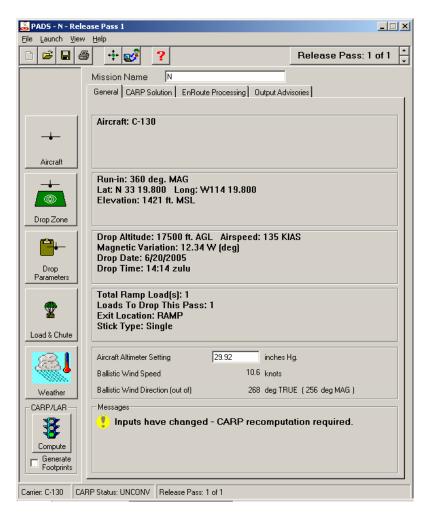


Figure 45. JPADS-MP main GUI.

In these examples, the *N* mission data already been entered. To begin anew, the user would need to enter a mission name and save the file. Clicking on the *Aircraft* tab allows the user to selects either the C-130 (the default) or the C-17. The next figure shows the *Drop Zone* tab:

Drop Zone - N - Release	Pass 1
⊢Point of Impact / D	Z Wind Forecast Reference Point
Coordinates	Latitude N/S Degrees Minutes
Select From Map	N 33 19.800 Longitude E/W Degrees Minutes W 114 19.800
Elevation	1421 👹 ft. (msl)
Approach	
Run-in	360 deg. MAG
	Close
Carrier: C-130 CARP Sta	tus: UNCONV Release Pass: 1 of 1

Figure 46. Drop Zone GUI

The *Drop Zone* tab allows the user to select the desired PI Wind Forecast Reference Point in terms of Latitude, Longitude, and Elevation. This is the point that the wind file will be centered on. For this study, the coordinates for Site 16 at YPG are used as this is the point from which the weather balloons were launched. This window is also where the aircraft approach data is defined. It is worth noting that the window is titled *Release Pass 1*. A key feature of JPADS is the ability to drop on multiple PIs or make multiple release passes. For this research, only a single Release Pass is used.

The next tab is Drop Parameters.

Drop Parameters - N - Release I	Pass 1			
Release Info				
Drop Altitude	17500	ft.	AGL	
Indicated Airspeed	135	KIAS		
Scheduled Drop Date	6 /20/2005 👻			
Scheduled Drop Time	14:14		zulu	
L Marrielle Casalle Dalassa Baint				
Manually Specify Release Point Release Point				
Coordinates Latitud	le Degrees Minutes			
₩75	Degrees Minutes			
Select From Map				
EAW Degrees Minutes				
Carrier: C-130 CARP Status: UNC	ONV Release Pas:	s: 1 of 1		

Figure 47. Drop Parameters GUI

This window allows the user to define details of airdrop in terms of Altitude, Airspeed, Date, and Time. The user has the option to manually specify a release point, but that option defeats the purpose of this research and is not used. It is critical for the data and time set here be in agreement with the date and time of the weather data the user intends to use for mission planning, otherwise the results will be invalid. The user must next fill out the *Load & Chute* tab. For full details of options for this page, please refer to the most current PSI published JPADS-MP user's manual.

Load & Chute - N - Release Pass 1		
Mission Data	Load Specific Data by Load Number	245
Total Ramp Loads (All Releases)	Load Number RAMP 1	257 27
Current Release Pass 1 of 1	Chute/System Type Screamer	7 297
Release Data	Map Label (Optional, 15 chars max) Release Delay (from previous numbered	317 3
Exit Location RAMP	load in this release) (Optional, sec)	1 357
Loads to Drop This Release	Total Rigged (All-up) Weight (lbs) 8000 Flight Station (load c.g.) 677	7
Stick Type Single	Stick Position	397
Glide Safety Factor 0.890	Guidance Unit/CNIU ID (Only required 1001	417 437
	for wireless communication)	457
	Latitude N/S Degrees Minutes	477
	Select From Map Longitude	497 5
	E/W Degrees Minutes	517
	W 114 22.226 PI Elevation (it. MSL) 1249 Image: March and a state of the	557-
	Ballistic Chute Type 2 G11	577 55 -
	Steerable Chute Type 850 SQ-FT	61 61
		637
		657 6
	Set Same As Previous	74-
	RAMP Load	
		167
		787
		07 827
		847-
	Close	869
Carrier: C-130 CARP Status: UNCONV Release Pass: 1 of 1		

Figure 48. Load & Chute GUI

For this research, the settings in *Load & Chute* tab are held constant. The guided parachute system is the Screamer with 850 ft² parachute. The Ballistic Chute Type is set to two G11 parachutes. The PI coordinates and elevation are set here. This representative mission was created using an actual test point from the ACTD program, only the Run-In heading was changed to a cardinal direction. As a result, the PI is set as being the JPADS Center PI target as used in testing. This is located 3.7 km from Site 16. It may have been better for the purposes of this analysis to have set Site 16 to be the PI. Unfortunately, this was realized too late for implementation. However, any error incurred by this is believed to be minimal when considering that the minimum tested weather resolution was 5 km. Total Rigged Weight is 8000 lbs at Flight Station 677 on the left side of the fuselage. CNUI ID is set as 1001, but is not needed. The Glide Safety Factor is left at the default setting of 0.89. Finally the user is ready to gather weather data with the *Weather* tab.

Weather: Default Options		×
WeatherAcquisition WindFileProduction Tools Help		
Drop Information	Wind File Production	
Planned Drop	Build Wind File	
Release Pass: 1 of 1	Build Status	
Point of Impact N 33 19.800 ₩ 114 19.800	Best Available:	
Drop Time : 06/20/2005 (zulu) 14:14	(Balloon only)	
Drop Altitude: fft MSL] 18921		
Weather Acquisition		
Weather Source Inventory Age (hh:mm)		
Dropsondes 0		
Pilot Report 0		
4D Forecast 0	L Full Options	

Figure 49. Weather GUI

The *Weather* tab is where most of the work in this research was done. The Drop Information shows data copied from the previous tabs. The next step is to acquire weather data. This is done by selecting one of the options under the Weather Acquisition section. The options relevant to this research are: Dropsondes, 4D Forecast, Balloon, and Climatology. In the interest of space, only the 4D Forecast method will be shown here. Other methods work similarly; where there are differences, they are detailed in the manual. Selecting an option under Weather Acquisition brings up the Weather Source – Acquire 4D Forecast GUI.

 Browse
Close
e Status

Figure 50. Weather Source GUI.

The JPADS-MP uses the 4D Forecasts generated by AFWA. These come in a format known as GRidded Information in Binary format (GRIB) files. Once these are downloaded, the *Browse* button is used to point the Weather Source GUI to the location of required GRIB files. Once the appropriate path is specified in the "GRIB Files Location" field, select the "Acquire Forecast" button. This will read the weather forecast into the JPADS-MP Environmental Data folder. Once this is complete, a *windgui* window opens showing the duration of valid times for the forecast as well as the forecast coverage area.

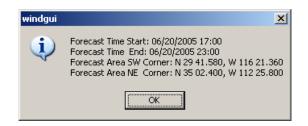


Figure 51. "windgui" Information Window

The user can then close this window as well as the Weather Source window and return to

the Weather tab.

🔣 Weather: Default Options	×
WeatherAcquisition WindFileProduction Tools Help	
Drop Information Wind File Production	
Planned Drop Build Wind File	
Release Pass: 1 of 1	atus —
Point of Impact N 33 19.800 Best Available:	
Drop Time : 06/20/2005 (LAPS Forecast-InSitu) Cance	el
Drop Altitude : frt MSL) 18921	
Weather Acquisition	
Weather Source Inventory Age (hh:mm)	
Dropsondes 0	
Pilot Report 0	
4D Forecast 1 00:00 Full Options	Close

Figure 52. Weather GUI with 4D Forecast loaded.

Here, the 4D Forecast inventory now shows an increment of one and the Wind File Production section now has the options for wind file generation via LAPS Forecast-only, Forecast Only, and Climatology only. These are listed in order of best to worst methods for calculating an accurate wind file. The Local Analysis and Prediction System (LAPS) is the most advanced modeling method included within the JPADS-MP. It allows complex modeling of wind interaction with terrain features such as how wind will flow over or around terrain obstacles. Select either the *Best Available* or LAPS Forecast-only buttons to begin Wind File production.

Weather: Default Options	×	Weather: Default Options	×
$\underline{W} eather Acquisition \underline{W} ind File Production \underline{T} cols \textbf{Help}$		$\underline{W} eather Acquisition \underline{W} indFileProduction \underline{T}ools \underline{H} elp$	
Drop Information	Wind File Production	Drop Information	Wind File Production
Planned Drop	Build Wind File	Planned Drop	Build Wind File
Release Pass: 1 of 1	Build Status	Release Pass: 1 of 1	Build Status
Point of Impact N 33 19.800 W 114 19.800	Best Available:	Point of Impact N 33 19.800 W 114 19.800	Best Available:
Drop Time : 06/20/2005 (zulu) 14:14	(LAPS Forecast-InSitu)	Drop Time : 06/20/2005 (zulu) 14:14	(LAPS Forecast-InSitu)
Drop Albitude : (ft MSL) 18921		Drop Alifude : (R MSL) 18921	
Weather Acquisition		Weather Acquisition	,
Weather Source Inventory Age (hh:mm)		Weather Source Inventory Age (hh:mm)	
Dropsondes 0		Dropsondes 0	
Pilot Report 0		Pilot Report 0	
4D Forecast 1 00:00	Full Options	4D Forecast 1 00:00	Full Options

Figure 53. Weather GUI during Wind File production and production complete.

The LAPS Wind File production takes a few minutes to run. During this time, the Build Status indicator will be yellow and display "In Progress". This will change to "Complete" and the status bar will be full once the LAPS Wind File has been generated. The "Weather" tab can now be closed to return to main JPADS-MP GUI page. Once here, the user selects the *Calculate* button under the CARP/LAR section. Unless FalconView is installed, do not select generate footprints as it will result in a crash of the JPADS-MP (this option was not used in this research).

e <u>L</u> aunch <u>V</u> ie	lease Pass 1 w <u>H</u> elp			
) 🖻 🖬 (s 🕂 💕 ?		Releas	se Pass: 1 of 1
	Mission Name N			
	General CARP Solution Er	nRoute Processing Output	Advisories	
	CARP	Nominal	Late	
	Latitude N 33 15.325	N 33 18.490	N 33 21.654	
→	Longitude W114 23.285	W114 22.457	W114 21.628	
	Short/Long Short 8880	Short 2308	Long 4265	Yards
Aircraft	Left/Right Right 104	Right 104	Right 104	Yards
<u> </u>	Altitude			
	TRUE (MSL)	18921	ft.	
	Pressure	17816	ft.	
Drop Zone	Indicated	17816	ft.	
m .	AGL	17500	ft.	
	D-Value	-1105	ft.	
Drop Parameters	CARP Info			
1 arameters	Convergence Status	CONV		
	Optimization	Not Applicable		
- 🏆	95% CEP	N/A	Yards	
	Footprint Generation Progress	0	200 Runs Max	
_oad & Chute	- Load Specific			
8]	RAMP Load 1			
Weather	Time of Fall From Green Light	(mm:ss) 03:46		
CARP/LAR	Messages			
	CARP computation completed successfully - No footprint generation requested.			
Generate Footprints				

Figure 54. JPADS Main GUI CARP Solution TAB after successful CARP calculation.

Selecting *Compute* CARP will automatically open the CARP solution tab. The CARP section shows the Latitude and Longitude of the Early, Nominal, and Late CARPs which also define the boundaries of the LAR. In order to collect this data, an Optical Screen Reader tool was developed by Captain Ryan Eggert of the Air Force Research Laboratory Advanced Architecture and Integration Branch.

💑 ScreenOCR			×
Mission Name N_R1.scm Additional Comments f	Specify File No output file specified or Capture File	Quit	
1414Z ☐ 5km Data ☐ 15km Data ☐ 45km Data ✔ Wx Balloon			

Figure 55. Screen OCR Main GUI

The Screen OCR tool reads the values in the Early, Nominal, and Late CARP coordinate boxes and copies them to a text file. In doing so, it also converts them from a DDD MM.mmm format to a DDD.dddddd format. The conversion to decimal degrees allows for easier mathematical operations later. Additionally, the Screen OCR copies the coordinates for the Screamer OP from its memory location and writes it to the same text file. To use the Screen OCR, ensure that the Mission Name is correct. This must match name of the Screamer mission data in memory. It will be mission followed by "_R1.scm". Check the type of data from which the wind profile is being generated and type in the launch time for a weather balloon (i.e., 1414Z) or the initialization time for a weather forecast (i.e., 0600 ini). Select "Specify File."

Save As				? ×
Save jn: 隘	20050620	• 🗢 主	📸 🎫	
File <u>n</u> ame:	20050620 1414Z Data.txt		<u>S</u> ave	
Save as <u>t</u> ype:		•	Canc	el

Figure 56. Screen OCR File Save As GUI

If starting a new file (for a weather balloon), enter the file name "YYYYMMDD TTTTZ Data.txt". Otherwise, click the appropriate existing file and select *Save*.

The method of building text files for analysis is to segregate the data by weather balloons. The Screen OCR allows for a new file to be opened and then to append subsequent data to this file. First, the CARP/LAR/OP is calculated for a weather balloon. This data is saved to a new file bearing the date and time of the balloon launch as the file name. Next the CARP/LAR/OP is calculated for each weather forecast that was valid for the time of that weather balloon launch. Each new data set is appended to the text file resulting in a file similar to the one shown below:

📕 20050620 1414Z Data	a.txt - Notepad						
<u>File E</u> dit F <u>o</u> rmat <u>V</u> iew	Help						
33.255633 33.259800 33.254900 33.259850 33.254900 33.263250 33.258950 33.258950 33.258417 33.255267	-114.387967 -114.37793 -114.380117 -114.377450 -114.380650 -114.379650 -114.379283 -114.380717	33.361117 33.365283 33.360383 33.365333 33.360383 33.368733 33.368733 33.364433 33.361900 33.360750	-114.360333 -114.35017 -114.352500 -114.349833 -114.352033 -114.352033 -114.352667 -114.351667 -114.353100	33.326595 33.327322 33.326797 33.327315 33.326780 33.327645 33.327645 33.327645 33.327645 33.327645 33.326815 33.326870	-114.371227 -114.370378 -114.370388 -114.370380 -114.370485 -114.37045 -114.370515 -114.370515 -114.370397	WX Balloon 15km Data 15km Data 5km Data 45km Data 45km Data 45km Data 45km Data	14142 0600 ini 1800 ini 0600 ini 1800 ini 1200 ini 1800 ini
•							

Figure 57. Sample text file record of CARP and OP calculations from the JPADS-MP captured by the Screen OCR program.

As can be seen, each line represents a different weather input: weather balloon on the first line, followed by weather forecasts of varying resolution and initialization time. The coordinates of the CARPs and OP are to the left of the metadata. Capt Eggert also developed a CARP Analysis tool to generate Northing and Easting data from the raw coordinates captured by the Screen OCR. Note that in the Screen OCR created text file the column contain information in the following order: Early Lat/Early Lon – Nom Lat/Nom Lon – Late Lat/Late Lon – OP Lat/OP Lon – Metadata However, the CARP Analysis tool changes this order to:

Nom NS/Nom EW – Early NS/Early EW – Late NS/Late EW – OP NS/OP EW - Metadata

🚣 CAR	P Analysis	×
	Generate Northing Easting	
	Quit	

Figure 58. CARP Analysis Tool Initialization GUI.

Selecting the CARP Analysis tool opens the GUI above. Selecting Generate Northing

Easting brings up the Open window for file selection.

Open	<u>?</u>]	×
Look in: 🗀	20050620 💌 🗧 🖆 🏢 🗸	
20050620	1414Z Data.txt	
File <u>n</u> ame:	20050620 1414Z Data.txt]
Files of type:	Text Files (*.txt) Cancel	
	Dpen as read-only	/

Figure 59. File selection GUI for CARP Analysis Tool.

After finding the correct file to analyze, select Open. This opens the Save window.

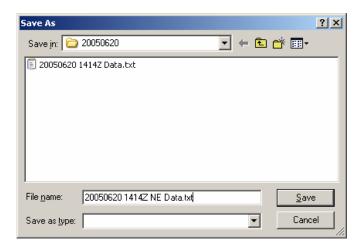
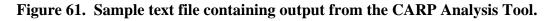


Figure 60. File Save As GUI for CARP Analysis Tool.

Be extremely careful to change the file name here, otherwise you will destroy you data as well a getting no results! Northing/Easting data in this study is noted by an "NE" in the file name to distinguish it from the raw data text file. The CARP analysis tool functions by comparing each weather forecast to the weather balloon data in line one of the text file. This results in a file similar to this:

) 2	0050620 1414	Z NE Data.txt - Nol	tepad						<u>- 0 ×</u>
File	<u>E</u> dit F <u>o</u> rmat	⊻iew <u>H</u> elp							
67 33 69 41 65 80 8 95	Early 953.5 731.4 979.9 681.8 774.9 998.6 809.1	EW La 87374 46 90987 -8 56292 46 24147 -8 33989 84 02194 36 92097 86	81.268824 67.752795 81.272429 44.825326	OP NS OP EW 950.764528 729.028579 977.194776 679.421502 772.419161 806.543093 673.182890	Source Lead-Ti 462.101579 -81.381192 467.649755 -81.384795 844.728703 367.831778 86.876783 -40.680933	me 79.047517 78.116919 82.585575 69.085537 98.878861 86.961669 66.292295 77.278890	80.632102 22.404192 79.855759 20.518648 116.456310 70.650243 24.400489 30.500636	15km Data 15km Data 5km Data 45km Data 45km Data 45km Data 45km Data	814 2014 814 2014 814 1414 814 2014 2014
4									▼ ■



In this file, the data represents error in the forecasting. A value in the Nominal NS column of -40.578024 means that particular forecast generated a Nominal CARP coordinate that was 40.578024 m South of the correct Nominal CARP coordinate as

defined by the Nominal CARP calculated from the weather balloon (an actual sampling of the atmosphere). Also note that, while the resolution data is unchanged, the initialization time has been replaced by the Lead-Time. This is accomplished by simply taking the difference between the weather balloon launch time and the forecast initialization time. The data from each weather balloon (and its corresponding forecasts) is saved in a folder named for the day the balloons were launched on.

Drop Date	# WxBs	WxB File	WxB Launch Time
20050620	2	08L	1414Z
		11L	1729Z
20050621	2	07L	1316Z
		10L	1557Z
20050624	3	07Z	1336Z
		08L	1458Z
		10L	1632Z
20050815	3	04L	1126Z
		08L	1403Z
		10L	1615Z
20050816	4	04L	1100Z
_		05L	1148Z
		07L	1350Z
		10L	1709Z
20050817	4	04L	1037Z
		05L	1127Z
		08L	1418Z
		11L	1757Z
20050818	3	03L	1008Z
		05L	1159Z
		09L	1549Z
20050819	3	04L	1053Z
		05L	1151Z
		08L	1441Z
20050912	4	04L	1114Z
		05L	1205Z
		06L	1255Z
		07L	1409Z
20050913	1	09L	1558Z
20050915	2	05L	1145Z
		07L	1332Z
20051019	2	04L	1130Z
		07L	1355Z
20051020	4	04L	1111Z
		06L	1303Z
		07L	1356Z
		11L	1813Z
20051021	2	04L	1104Z
		06L	1257Z
20060125	1	06L	1234Z
20060126	2	10L	1654Z

Appendix H: Weather Balloon Data Listing

		12L	1846Z
20060227	2	09L	1541Z
		12L	1911Z
20060228	2	08L	1529Z
		11L	1804Z
20060301	3	10L	1705Z
		12L	1836Z
		14L	2046Z
20060303	2	06L	1259Z
		09L	1532Z
20060327	3	08L	1431Z
		12L	1842Z
		13L	2002Z
20060328	3	09L	1540Z
		11L	1739Z
		12L	1854Z
20060329	1	11L	1731Z
20060330	3	06L	1254Z
		09L	1544Z
		11L	1817Z
20060508	2	09L	1539Z
		11L	1825Z
20060509	2	08L	1440Z
		10L	1638Z
20060510	2	06L	1306Z
		08L	1518Z
20060511	2	08L	1434Z
		12L	1836Z
20060613	2	07L	1339Z
		08L	1450Z
20060614	2	07L	1348Z
		12L	1830Z
20060615	2	07L	1342Z
		08L	1437Z
20060616	1	08L	1438Z
20060725	1	09L	1533Z
20060726	2	06L	1241Z
		08L	1432Z
20060727	2	06L	1304Z
		07L	1426Z
20060728	1	08L	1457Z
20060911	1	08L	1444Z
20060912	1	06L	1256Z

20060914	2	06L	1314Z
20000011		07L	1427Z
20060915	2	06L	1252Z
		09L	1602Z
20061016	3	07L	1417Z
	-	09L	1533Z
		10L	1714Z
20061017	3	07L	1337Z
		08L	1517Z
		10L	1651Z
20061018	2	07L	1340Z
		08L	1509Z
20061019	3	07L	1412Z
		09L	1541Z
		12L	1913Z
20061020	2	08L	1433Z
		09L	1557Z
20061127	4	05L	1132Z
		11L	1812Z
		13L	1950Z
		15L	2141Z
20061128	2	07L	1409Z
		09L	1534Z
20061129	1	12L	1923Z
20061130	3	07L	1406Z
		08L	1505Z
		11L	1817Z
20061201	3	08L	1444Z
		09L	1608Z
		10L	1707Z
20061204	1	05L	1133Z
20061205	3	07L	1420Z
		09L	1549Z
		11L	1816Z

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 14. ABSTRACT The United States Air Force is partnering with the United States Army as well as allied nations to develop a revolutionary advance in logistical support known as the Joint Precision Air Drop System (JPADS). The focus of this study is to develop a process to quantitatively analyze system sensitivities to various types of weather inputs and the corresponding effect on system accuracy. Weather balloons were used to provide representative "truth" to which forecast weather could be compared. Each data type was fed into the JPADS Mission Planner to produce navigation points which could then be compared statistically. The process was tested on a limited data set to provide a first look at the variables of forecast resolution and "lead-time." Initial results indicate best system accuracy is achieved for lowest forecast resolution (i.e., 45 km vs. 5 km data) and shortest lead-time (i.e., <12 hrs vs. >12 hrs). This result will not only allow for better accuracy of JPADS, but also reduce bandwidth and transmission time necessary to send weather forecast data to the warfighter. 15. SUBJECT TERMS						
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